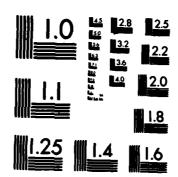
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'OPY RESOLUTION TEST CHART



NAPC-PE-94-C

AD-A146 544

EFFECT OF ASPERITY INTERACTIONS ON SCUFFING

JOHN I. Mc COOL

SKF TECHNOLOGY SERVICES SKF INDUSTRIES, INC. 1100 FIRST AVENUE KING OF PRUSSIA, PA 19406-1352

MAY 1984

FINAL REPORT

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PREPARED FOR:

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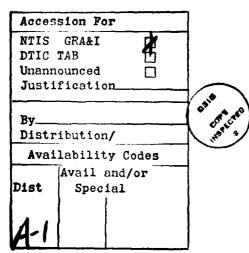
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1.0 INTRODUCTION AND SUMMARY

Scuffing

Scuffing, smearing, and scoring are synonyms for a failure mode that affects both gears and rolling contact bearings. It is characterized by an abrupt increase in friction and rapid metal transfer. Once begun it will, if not arrested by a load reduction or decrease in sliding velocity, rapidly become self aggravating due to interfacial metal plowing. It is usually therefore considered a catastrophic failure mode.

It is generally believed that scuffing either will or will not occur depending for a given set of conditions on the magnitude of the load. Interest has, therefore, focused on determining the 'scuffing' load as a function of the many variables that characterize a rolling/sliding contact.

The most widely used criterion for linking scuffing load and the contact variables is one based on the conjunction temperature, i.e. the sum of the bulk temperature of the contacting bodies and the 'flash temperature' i.e. the temperature elevation due to friction in the contact zone. With the flash temperature computed via a thermal analysis, the conjunction temperature criterion links scuffing load to the thermal conductivity, specific heat and density of the contacting bodies, the friction coefficient, the load and the surface velocities of the contacting

bodies. It does not specifically link scuffing load to microgeometry though widespread experience suggests that microgeometry exerts an influence.

The objective of the investigation reported herein was to determine whether the characteristics of surface asperity interaction can be used to refine predictions of the load at which lubricated surfaces undergoing relative rolling and sliding, will scuff or smear.

In this investigation a series of tests were run in a geared roller test machine using three combinations of rolling and sliding velocities, and for each velocity condition, a set of polished and ground rollers. A description of the test method is given in Appendix VIII. For the surfaces and speeds used, the film parameter i.e. the ratio of the film thickness h to the mean square roughness o, ranged from 0.082 to 1.01. The load was increased until the occurrence of scuffing. Each test was then repeated until the load reached half the scuffing load.

For the same set of velocity conditions the polished specimen pair invariably scuffed at a higher load than the ground pair.

The surfaces were traced using a stylus profile instrument and the output signal digitized and processed to yield mean

square values of the profile height, slope and curvature. The computerized system for acquiring and processing such data is described herein along with the associated calibration and system checkout procedure.

The measured surface characteristics were used as input to a simulation model, ASPERSIM, which computes the expected number of micro-contacts and the expected force and area of each.

Significance tests showed that the polished specimens had lower average asperity force and average asperity area than the ground specimens and that these values were correspondingly lower for the unscuffed than for the scuffed tests.

For the specimens that were run to scuffing, (both ground and polished) average asperity area increased with scuffing load. For the polished specimens average force also increased with scuffing load. This is believed to reflect the alteration of microgeometry with load and not a causal relationship between these variables and scuffing load.

Corresponding significance tests for conjunction temperature failed to distinguish a difference due to finish type. Tests using the film parameter i.e. the ratio of film thickness to surface roughness showed a weakly significant effect of finish and scuffing status.

An evaluation using unrun microgeometry showed the polished specimens to have a lower average asperity force than the ground specimens and a lower total expected contact density. It is suggested that the product of average force and expected contact density may be a useful inverse measure of proneness to scuff.

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2.0 THEORETICAL BACKGROUND

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The following discussion outlines the relevant background that formed the basis for undertaking the investigation reported herein. The scope of the investigation itself is outlined in Section 3.0.

2.1 Constant Conjunction Temperature Criterion

Blok [1] postulated that scuffing occurs when the maximum conjunction temperature reaches a critical value that depends on the specific combination of material and lubricant. Blok expresses the maximum conjunction temperature, $T_{\rm c}$, as the sum of the bulk temperature $T_{\rm b}$ and the maximum 'flash' temperature $T_{\rm f}$

$$T_c = T_b + T_f$$

For line contacts the maximum flash temperature is expressed under certain assumptions as follows:

$$T_f = 1.11 \text{ f W} | V_1^{\frac{1}{2}} - V_2^{\frac{1}{2}} | /bw^{\frac{1}{2}}$$

where

- w = width of contact
- b = $(k \rho c)^{1/2}$ a thermal factor evaluated for the material of the contacting bodies
- k = thermal conductivity

- c = specific heat per unit mass
- ρ = density
- V_1, V_2 = velocities of bodies 1 and 2
 - W = load/unit length (For elliptical contacts of length 2a and subject to load P, the equivalent load per unit length is taken as 3 P/4a).
- f = friction coefficient
 The assumptions involved in deriving this expression for the
 flash temperature include:
 - 1) The interfacial temperature is the same for both bodies.
 - 2) Heat is transferred by conduction perpendicular to the contact.
 - 3) The heat flux is semi-elliptically distributed across the contact.

There is no unanimity as to the physical mechanism by which smearing occurs. Explanations that have been proposed fall into these classifications:

1) Loss of the protective capacity of the oil due to an adverse change in the oil as a material. For example, the oil might lose its capacity for removing heat or for inhibiting welding.

- 2) Loss of the protective capacity of the oil as a structural element. Blok's thermal collapse model is a case in point. [Blok [2]].
- 3) Loss of the protective capacity of the boundary and oxide layers on the rubbing surfaces. The thermo-elastic instability mechanism discussed by Dow and Burton [3] could cause the destruction of boundary layers. Desorption of the boundary layers is discussed by Begelinger and DeGee [4].

2.2 Effect of Surface Roughness

The surface roughness does not appear explicitly in calculating the conjunction temperature. In recent work, however, Durkee and Cheng [6] present data showing differences in scuffing load of a factor of 15 between rough and smooth roller/ring pairs in simple sliding experiments conducted with 52100 material parts and lubricated with an additive-free napthenic oil.

To accommodate the effect of roughness in scuffing prediction is it necessary to use the somewhat artificial device of letting

Winer [5] has, however, proposed a flash temperature correction factor of $[1-0.7\sigma]^{-1}$ ($\sigma = CLA$ roughness in μ m).

the critical conjunction temperature depend on surface finish as well as the specific material and lubricant combination.

The problem with this approach, however, is that testing must be conducted using a range of surface finishes and finishing processes. What is preferable is to develop a means of directly accounting for the effect of surface finish in predicting scuffing load.

The effect of surface roughness on scuffing load must be due to the ways in which the existence of the roughness causes the interfacial conditions to depart from that assumed in Blok's analysis. The most obvious feature of rough contact is that the load is supported by a large number of asperities rather than uniformly over the contact.

Each asperity contact within the conjunction will support a different load depending on the asperity height and local geometry. The elastically deformed asperity contact area will likewise vary randomly from asperity-to-asperity.

The smearing load is thus likely to be influenced by these functional effects of surface roughness:

1) The expected number of asperity contacts within the conjunction

- 2) the expected asperity contact area and
- 3) The expected asperity load.

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These quantities depend on the statistical microgeometry of the surfaces and i) the applied load for dry contact and ii) the film thickness for EHD contact.

2.3 Calculating Asperity Interaction Behavior

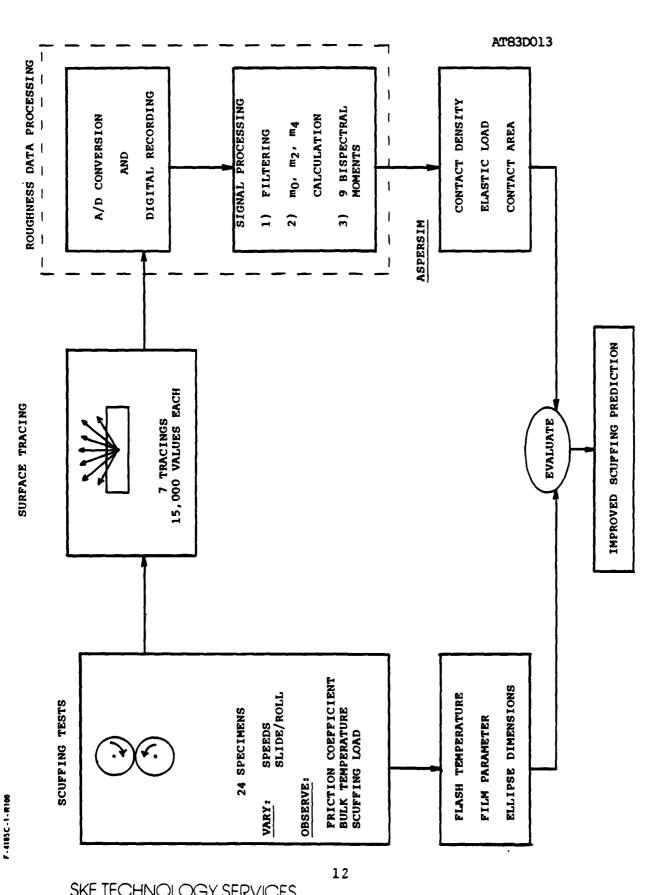
Calculating the expected number of asperities and their contact area and load for a general anisotropic rough surface is possible via the simulation computer program "ASPERSIM" developed at SKF under the sponsorship of the Department of Energy [7]. The input to the program comprises nine quantities known as bispectral moments which describe the height distribution of a surface as well as its frequency and anisotropic character. These nine quantities may be deduced [8] by multiple regression analysis from the mean square of the surface slope and the surface second derivative, calculated from a number of non-parallel profile traces. A detailed description is given in Appendix I.

Signal processing involves digitization of the output of a profile measurement device followed by filtering, editing, and mean square computation. Appropriate software for this digital

signal processing has also been developed at SKF and is discussed in Section 4.0.

3.0 PROGRAM SCOPE

The scope of the program reported herein is shown diagrammatically in Figure 1. Scuffing tests were conducted at the Naval Air Propulsion Center (NAPC) using a geared roller test machine. In these tests annular specimens having an outer diameter of 3.01" and a 14" lateral crown radius are run against each other in lubricated contact at independently set rotational speeds. The radially applied load is increased by 50 lbs every 5 minutes until scuffing occurs. The occurrence of scuffing is indicated by a rapid increase in surface temperature and torque. Response variables, in addition to the load at which scuffing occurs, are the friction coefficient and the bulk temperature of In the present test program the discs were made the specimens. of AISI 4720 steel. The test lubricant was a basestock pentaerythritol ester commonly used in formulating MIL-L-23699 lubricants. Half of the specimens had a ground finish on the order of 25μ in. RMS and half had been given an additional polishing operation resulting in a finish on the order of 10μ in. RMS. Three rolling/sliding speed combinations were selected to be run for each surface finish type. The test plan called for running a pair of rings until scuffing occurred at some load P and then to run an identical second pair until a load of half this value (P/2) was reached. The total number of discs is thus



SCOPE OF SCUFFING INVESTIGATION

FIGURE 1:

The second second

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24 corresponding to 3 speed combinations x 2 surface finishes x 2 loads x 2 discs per test.

For each test specimen, 7 surface tracings were initially made using the stylus system described in Section 4.0. The traces were made in the run-in area but not within the scuffed zone and at angular spacings of 22.5° but omitting the circumferential direction.

The analog signal output of each stylus trace was amplified, sampled, digitized and recorded on floppy discs.

Following data acquisition the files of digitized data were read and processed to yield the values of the spectral moments m_0 , m_2 , m_4 with and without digital filtering between a prescribed set of filter limits.

Using the measured spectral moment values on each of 7 traces, a multiple regression analysis program was used to deduce the 9 bispectral moments which characterize a general anisotropic surface (See Appendix I).

These 9 bispectral moments were to be input to computer program ASPERSIM, for computing the contact density, elastically supported load, and the true area of contact under the assumptions that surface asperities are described as second order func-

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tions and that the separation of the surfaces is equal to the calculated elastohydrodynamic lubricant film separating the surfaces at the velocity and temperature of each test. Because of irregularities in some of the initially processed data indicative of extraneous noise, data acquisition was repeated using a single axial trace and the equivalent values of the 9 spectral moments were deduced under the assumption of isotropic bodies. These values were used as ASPERSIM input.

The ASPERSIM output, together with the conventionally computed flash temperature, calculated film parameter, and ellipse dimensions were then examined and evaluated for evidence of an effect of surface roughness on scuffing.

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4.0 SURFACE DATA PROCESSING

4.1 Data Acquisition System

Figure 2 is a schematic representation of the data acquisition system used in this investigation. At the heart of the system is the Talysurf 4 surface profile measurement system. In operation, the Talysurf 4 causes a contacting stylus to move across the surface of interest and a voltage proportional to the stylus' excursions is produced. In the conventional mode of operation this voltage is filtered and used in a computation of the CLA average value of the roughness. In the present application the unfiltered voltage is fed into an amplifier and then into the A/D converter that operates in conjunction with a Digital Equipment Corporation PDP 11/03. In order to minimize electrical noise problems, the lowest possible stylus traverse velocity of 0.000466 in/sec is used to assure that the roughness signals correspond to frequencies well below 60 Hz.

4.2 Data Acquisition Software and Calibration

The data is acquired and stored on a floppy disc under the control of the computer program entitled "GETAD".

A system typewriter controls the input/output operations (I/O). Prior to the acquisition of data a calibration constant

FIGURE 2: ROUGHNESS DATA ACQUISITION SYSTEM

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is computed that relates, for the magnification setting at which the Talysurf is being used, the number of digitized integer values of the signal to the surface displacement. In performing the calibration constant calculation, the user is directed under the control of the program GETAD to place the stylus in contact with the lower level of a calibration step. In this program two Johansson blocks having a height difference of 300 µin. was used for data taken with a Talysurf IV vertical magnification setting of 5000X. The stylus remains in steady contact with the lower of the pair of blocks until 200 digitized values are read. user is then directed to place the stylus in contact with the higher level of the pair of blocks and an additional 200 digitized values are read. The program then averages the digitized values corresponding to the lower and higher levels of the step and asks the user to supply the actual height difference of the pair of blocks. The step height is then divided by the difference between the average digitized height values to yield the number of linear units (microinches) corresponding to a unit change in the digitized value of the surface. This ratio is termed the calibration factor.

The calibration factor and a user specified 6 digit file name are stored in a "header" block preceding the digitized profile data. The user of the system specifies the total number of

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points to be digitized and the frequency with which the points are to be taken. These two values together determine the total length of trace.

4.3 Processing Software

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The digitized profile data is re-read from the disc and processed by means of another program called "PRODOE". PRODOE reads the header information including the calibration factor and calculates the grand average of the digitized values. It then computes the distance, in digitized units, of each data point from the grand mean and scales it, by means of the calibration factor, into length equivalent units. The mean square surface height (m_0) , the mean square slope (m_2) and the mean squared value of the second derivative of the surface (m_4) are computed. m_4 is, to a good approximation, the mean square curvature of the surface.

PRODOE performs an editing function to remove outliers in the data record. For this purpose it uses either a user specified criterion or a computed criterion, being a multiple of the RMS value of the unfiltered surface to determine whether the difference between successive pairs of values could be real. If the criterion is exceeded, the second value is equated to the first. The system reports the number of such replacements that took place.

The user specifies a frequency band f_1 and f_2 in units of cycles per linear distance and the data are then digitally filtered using a cascaded Butterworth filtering scheme described in [9]. PRODOE then computes the values of m_0 and m_2 and m_4 using the filtered signal. It also compiles a histogram of the surface height distribution using a sub-sample of 300 points.

4.4 Validation of Acquisition and Processing Systems

Figures 3-5 show the printed record of the use of program GETAD followed by PRODOE to 1) digitize and record a sine wave of nominal five cycles per second frequency produced by a signal generator and 2) process it through program PRODOE. User input and responses are shown underlined. The dialogue is self explanatory. A total of 5,000 points were sampled and the sampling frequency used was 40 points per second or 8 points per cycle. By entering the stylus traverse velocity of 0.000466" per second, the 5 Hz sine wave became the conceptual equivalent of a sinusoidal surface having a spatial frequency of

f = 5 cycles/sec/.000466 in/sec = 10729 cycles/inch

The calibration factor was input from the keyboard as unity so that values subsequently displayed are the 'as digitized' numerical values. The output under the heading unfiltered,

Ľ

R GETAD

ENTER NAME OF ANALOG SOURCE (UP TO 20 CHARACTERS)

SIGGEN

WILL ALL RUNS COLLECT EQUAL NUMBERS OF DATA POINTS?(Y OR N) WILL ALL RUNS COLLECT DATA AT THE SAME FREQUENCY?(Y OR N)

HOW MANY DATA POINTS? (<=60160) USE DECIMAL POINT IF ENTRY > 32767

5000

WHAT FREQUENCY (HZ<=1.E6)?

USE DECIMAL POINT IF ENTRY > 32767 40

.000466

CURRENT CALIBRATION FACTOR = 0.0000E+00 MEASURED UNITS / DIGITIZED UNIT IS THIS SATISFACTORY? (Y OR N)

WILL FACTOR BE ENTERED FROM KEYBOARD (K) OR CALCULATED WITH AID OF A/D BOARD (C)?

WHAT IS THE STYLUS TRAVERSE VELOCITY (IN/SEC)?

ENTER CALIBRATION FACTOR AS MEASURED UNITS/DIGITIZED UNIT = USE DECIMAL POINT IF VALUE LARGER THAN 32767

ENTER FILE NAME (UP TO 6 CHARACTERS) SI5000

FIGURE 3: SINE WAVE PROCESSING RECORD

START SIGGEN-THEN PRESS RETURN
THIS WILL START DATA COLLECTION

STOP SIGGEN

ANOTHER RUN? (Y OR N) N

STOP --

.R PRODOE

ENTER FILE NAME (UP TO 6 CHARACTERS) SI5000

CALCULATING AVERAGE OF ALL POINTS ************* (19 PASSES)

CALCULATIONS FOR UNFILTERED-UNSMOOTHED DATA ************ (19 PASSES)

UNFILTERED, UNSMOOTHED DATA

AVE = 2.75127E+02 U-INCH

MO = 8.33254E+05 U-INCH**2

M2 = 4.24161E+03

M4 = 2.16341E+01

CURRENT VALUE OF RES IS 0.38728E+04 IS THIS SATISFACTORY? (Y OR N) \underline{Y} CALCULATIONS FOR SMOOTHING DATA ****************** (19 PASSES)

CURRENT FREQUENCY CUTOFF VALUES ARE :

LOWER CUTOFF = 0.330000D+02

UPPER CUTOFF = 0.320000D+05

ARE THESE SATISFACTORY? (Y OR N) Y

CALCULATIONS FOR FILTERED-SMOOTHED DATA *********** (19 PASSES)

DOE DIGITIZED SURFACE DATA

TOTAL NO. OF SURFACE ELEMENTS = 5000.

F1 - LOWER FREQ. BAND CUTOFF = 3.30000E+01 CYCLES/INCH

F2 - UPPER FREQ. BAND CUTOFF = 3.20000E+04 CYCLES/INCH

DX - SURFACE INCREMENT = 1.16500E-05 INCHES

CALIB - CALIBRATION FACTOR = 1.00000E+00 U-INCH/COUNT

FIGURE 4: SINE WAVE PROCESSING RECORD (CONTINUED)

M2 = 4.23894E+03

M4 = 2.16090E+01

[FLAG = 0

HISTOGRAM OF FILE DK SI50000FF

MAX = 0.13177E+04 MIN = -0.13641E+04

TOTAL NUMBER OF POINTS = 294.

UPPER CELL BOUNDARY	NORMALIZED CUMULATIVE FREQUENCY	POINT FREQUE	NCY .
-0.11964E+04	0.1293	38	XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX
-0.10288E+04	0.2143	25	XXXXXXXXXXXXXXXXXXXXXXX
-0.86122E+03	0.2721	17	XXXXXXXXXXXXXXX
-0.69361E+03	0.3197	14	XXXXXXXXXXXX
-0.52600E+03	0.3605	12	XXXXXXXXXXX
-0.35839E+03	0.4014	12	XXXXXXXXXXX
-0.19078E+03	0.4524	15	XXXXXXXXXXXXX
-0.23171E+02	0.4932	12	XXXXXXXXXXX
0.14444E+03	0.5340	12	XXXXXXXXXXX
0.31205E+03	0.5816	14	XXXXXXXXXXXX
0.47966E+03	0.6156	10	XXXXXXXXX
0.64727E+03	0.6735	17	XXXXXXXXXXXXXXX
0.81488E+03	0.7211	14	XXXXXXXXXXXXX
0.98249E+03	0.7823	18	XXXXXXXXXXXXXXXX
0.11501E+04	0.8605	23	XXXXXXXXXXXXXXXXXXXXXXX
0.13177E+04	1.0000	41	XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX

ANOTHER RUN? (Y DR N)

FIGURE 5: SINE WAVE PROCESSING RECORD (CONTINUED)

unsmoothed data shows the average expressed in length units. It is positive indicating some small positive bias in the signal generator output. The quantity RES is used for editing or smoothing wild points. We have selected it to be 3 standard deviations of the difference $x_1 - x_2$ of two independent random values. Since $\sigma^2 x_1 - x_2 = \sigma^2 x_1 + \sigma^2 x_2$, $3\sigma_{x_1 - x_2}$ becomes, if x_1 and x_2 have the same variance $= 3 \bullet \sqrt{2} \bullet \sigma_x = 4.24\sigma_x = 4.24 m_0^{1/2}$. The default frequency band pass is $f_1 = 33$ cycles/in. to $f_2 = 32,000$ cycles/in. Since the spatial equivalent of the sine wave used here was 10,729 cycles/in., it fell well within this pass band and the m_0 , m_2 , and m_4 values after filtering differ little from the unfiltered values. The variable IFLAG denotes the number of points that were edited using the RES criterion. The histogram shows the typical "U" shape characteristic of sinusoids.

To determine the plausibility of the computed spectral moments they were compared to the theoretical values for a sine wave. A sine wave of amplitude A and spatial frequency f has the following theoretical values of m_0 , m_2 and m_4 :

$$m_0 = A^2/2$$

 $m_2 = (2\pi f A)^2/2$
 $m_4 = (2\pi f)^4 A^2/2$

From the maximum and minimum values printed just above the histogram the amplitude is deduced to be A = 1341 giving

$$m_0 = (1341)^2/2 = 8.99E5 \text{ vs.} 8.33E5 (µin)^2$$

For the nominal frequency of 5 cycles/sec = 10729 cycles/in.

$$m_2 = (2\pi \cdot 10729E6 \cdot 1341)^2/2 = 4.09E3 \text{ vs. } 4.24E3$$

and
$$m_4 = (2\pi \cdot 10729E-6)^4 (1341)^2 = 18.5 \text{ vs. } 21.6 (\mu \text{in})^{-2}$$

It is seen that the computed quantities are all of the correct order of magnitude, but differ from the theoretical values by up to 14% (m_4) . The actual error is not believed to be this great. The amplitude used in the above calculations was determined from just two data values and is subject to error due to noise. The frequency was also only nominal. As a further check we determined the A and f values implicit in the measured m_0 and m_2 and used them to predict m_4 . The results are

$$m_0 = A^2/2 = 8.33254E5$$

A = 1290 µin.

$$m_2 = (2\pi f \cdot 1290E-6)^2/2 = 4.24161E3$$

 $f = 11363$

so that my becomes,

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$$m_{4} = \frac{(2\pi \bullet 11363E-6)^{4}(1341)^{2}}{2} = 23.4$$

which is an 8% error.

5.0 TRACING PARAMETERS

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In processing roughness data it is necessary to specify the sampling interval Δx and the total length of trace L or, alternatively, the sampling frequency f_S (samples/sec) and the total number of sampled values N. These quantities are related as follows:

 $L = N\Delta x$

 $N = f_S/V \bullet L$

so that, $\Delta x = V/f_s$

where V is the stylus traverse velocity. In the initial tracing work in this program a spacing of 14.5 µin was selected. This value is comparable to that used in other profile digitization studies in the literature (cf. Sidik and Coy [10]). A total of N = 15000 points was selected giving a total record length of

 $L = 15000 \bullet 14.5E-6 = 0.22 in.$

The sampling frequency was

 $f_S = V/\Delta x = 0.000466 \text{ in/sec/14.5E-6} = 32.2 \text{ samples/sec}$

In processing the data the initial band pass chosen for digital filtering was $f_1 = 33$ cycles/in. and $f_2 = 10,000$ cyles/in. The lower frequency corresponds to the usual filter setting for precision specimens recommended by the Talysurf supplier. The upper limit is believed to be well below the frequency at which electrical noise exceeds the signal level (cf. Church [11]). The upper limit was further refined in subsequent processing as discussed in Section 7.0.

6.0 SCUFFING TEST RESULTS

Table 1 summarizes the results of the scuffing tests conducted at NAPC. The tests numbered 1A through 6A, shown in Column 1, were the tests which proceeded to scuffing as indicated in Column 3. Tests 1B to 6B were conducted at the corresponding conditions to Tests 1A through 6A but proceeded only until a load equal to one-half the associated scuffing load was obtained. is seen that consecutive pairs of tests in each series (scuffed and unscuffed) were conducted at identical rolling and sliding speeds but on two surface finish types, namely, ground and polished. Sliding and rolling velocity are indicated in Columns 4 and 5 and the linear speeds of the faster and slower discs in each test are given in Columns 6 and 7. The load at scuffing, or in the case of Tests 1B to 6B, the load at test termination is given in Column 8. In calculating film thickness the oil viscosity should ideally be evaluated at the temperature of the thin layer of oil in the contact inlet. However, this temperature and hence the film thickness varies throughout the duration of the test. At startup, when the discs are cold, the oil inlet temperature is well aproximated by the oil supply temperature (180°F). At the highest load attained in the test, the relevant temperature is somewhere between the bulk temperature and the conjunction temperature.

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TABIL	LADLE

CONJUNCTION TEMPERATURE TR-°F	457	425	415	425	367	491	341	312	392	350	306	326
BULK TEMPERATURE T _B -°F	280	276	303	310	291	400	232	221	282	952	252	592
FLASH TEMPERATURE T _f -°F	111	149	112	115	26	91	109	91	110	\$	\$	57
FRICTION COEFFICIENT F	0.047	0.036	0.034	0.034	0.025	0.025	0.041	0.031	0.047	0.039	0.025	0.022
KNESS fn) T=T _C	3.95	4.27	3.76	3.65	7.38	5.24	5.88	6.47	4.25	4.86	9.99	8.89
LOAD FILM THICKNESS P h (µ in) 1bs) T=180°F T=T _C	13.4	13.2	11.3	11.3	19.1	19.1	14.0	13.8	11.8	11.8	20.02	20.0
LOAD F	996	9911	3516	3716	1466	2116	483	583	1758	1858	733	1058
v ₂ tn/sec)	276	9/2	276	276	299	295	276	9/2	276	276	552	552
V ₁ V ₂ LOAD (in/sec) (ibs)	414	414	336	336	673	673	414	414	336	336	673	673
ROLLING VELOCITY VR(1n/sec)	345	345	302	305	610	610	345	345	305	305	919	610
SCUFFED SLIDING ROLLING ? VELOCITY VELOCITY V _S (in/sec) V _R (in/sec)	138	138	8	8	121	121	138	138	9	8	121	121
SCUFFED	>	>	>	>-	>	>	2	z	*	×	2	=
FINISH	ی	۵	g	۰	G	•	ပ	•	ம	•	g	۵
1EST 180.	*	Z	38	\$	SA.	Y 9	18	28	38	48	58	89

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Column 9 gives the film thickness calculated at the oil supply temperature and at the conjunction temperature using Hamrock and Dowson's formula for fully flooded elliptical EHD contacts. A thermal correction for inlet heating due to Murch and Wilson [15] has been applied and resulted in an approximate 10% reduction from isothermal conditions. Column 10 contains the friction coefficient measured in the tests, and Column 11 the computed flash temperature. Column 12 is the observed bulk temperature and Column 13 the sum of the flash and bulk temperatures. In computing flash temperature it was necessary to compensate for the fact that the contact was elliptical and not rectangular as assumed in Blok's formula. This was done following Kelley [22] by using an equivalent unit load W_e = 3P/4a, where P = scuffing load and a = half width of the elliptical contact.

Reviewing the data in Table 1 it is seen that in general for the Tests 1A through 6A for given rolling and sliding velocities the scuffing load is consistently higher in the polished, as opposed to the ground tests. This is shown in the plotted Figure 6 showing scuffing load against velocity 'condition'. This difference, however, is not, particularly in Tests 3A and 4A, percentually large. Repeated tests at each condition would have been useful to gauge whether the smearing load differences bet-

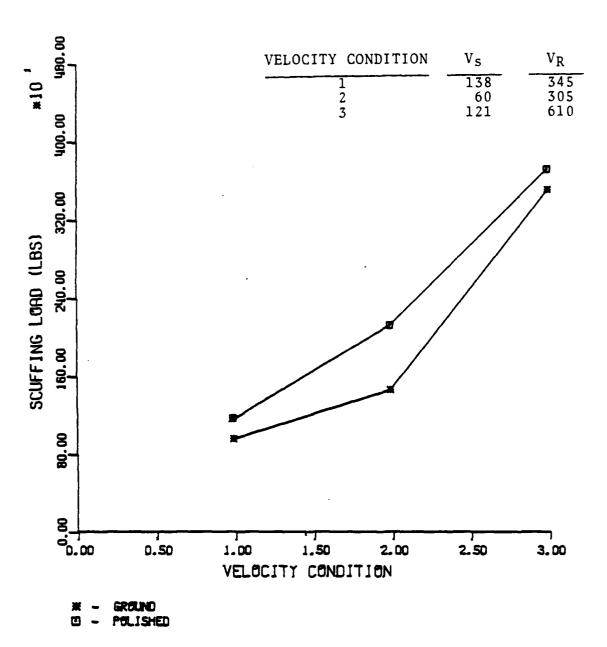


FIGURE 6: SCUFFING LOAD VS. VELOCITY CONDITION

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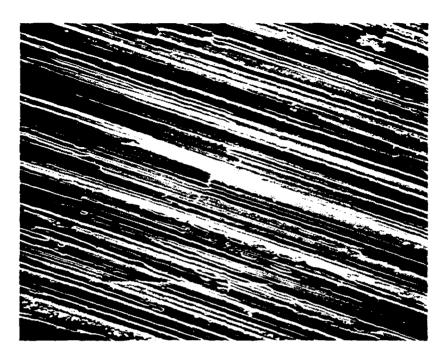
ween polished and ground are real. We have assumed them to be so. Tests 3A and 4A exhibited the highest scuffing load and had the lowest sliding speed. Tests 1A and 2A had the lowest scuffing load and the highest slide to roll ratio. Tests 5A and 6A had the same slide to roll ratio as Tests 3A and 4A but the sliding and rolling magnitudes were double those of Tests 3A and 4A. Since the observed scuffing load in Tests 5A and 6A are substantially lower than the value in Tests 3A and 4A, it is clear that scuffing is affected not just by the ratio of sliding to rolling but by their absolute values. Tests 1A and 2A have a comparable rolling velocity to Tests 3A and 4A but somewhat more than twice the sliding velocity. Here again, the scuffing load in Tests 1A and 2A is substantially below that obtained in Tests 3A and 4A. This indicates the important effect of sliding velocity on scuffing load.

Figure 7 shows comparative SEM photographs of the unrun ground and polished specimens. The surfaces are seen to be characteristically distinct with strong directionality evident in both. The ground surface exhibits very marked furrowing and the polished surface shows traces of where the furrows have been abraded away by the polishing operation.

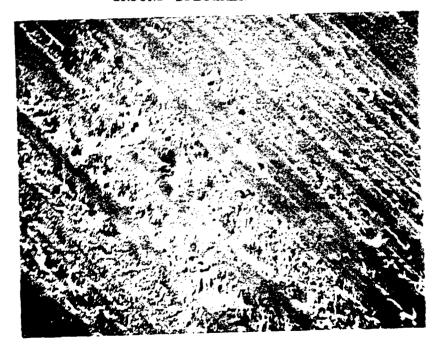
Figure 8 shows SEM photographs at 400X and 1000X within the scuffed area of the faster running of the two discs that ran

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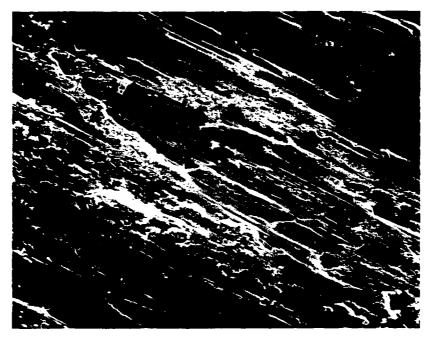
Q



GROUND SPECIMEN - 1000X



POLISHED SPECIMEN - 1000X
FIGURE 7: SEM PHOTOS OF UNRUN SURFACES



400X

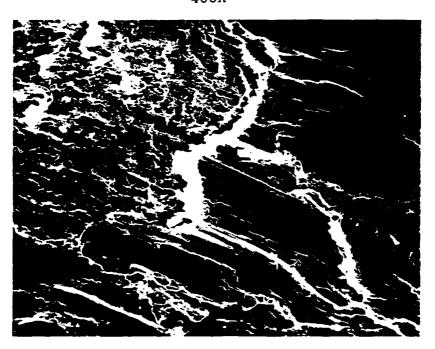
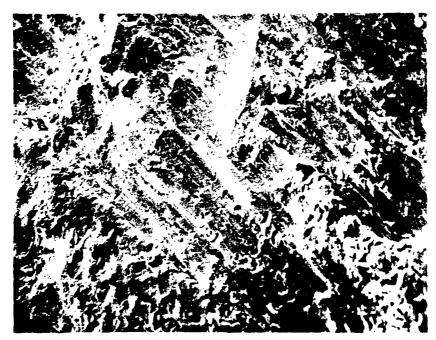


FIGURE 8: SCUFFED GROUND SURFACE
FAST DISC TEST 1A
SCUFFING LOAD = 966 LBS.

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in Test 1A. This test had the lowest observed smearing load and severe damage to the surfaces is quite evident in the photographs. Some visual evidence of directionality still persists despite the severe working of the surface. Figure 9 is a comparable set of photographs taken at 500% and 1000% in the scuffed area of the polished disc that was used as the faster running disc in Test 2A. This test had the second lowest scuffing load, 1166 lbs, observed during the test program. Again, evidence of gross damage is abundant but considerably less dramatic than the severe surface damage seen in the ground surface photographs.



500x

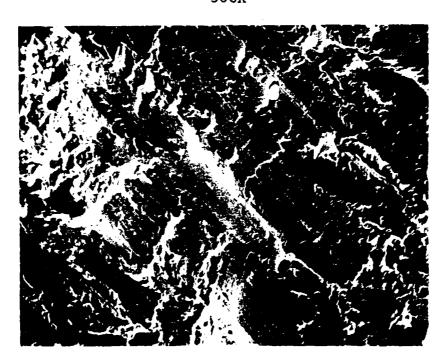


FIGURE 9: SCUFFED POLISHED SURFACE
FAST DISC TEST 2A
SCUFFING LOAD = 1166 LBS.

7.0 SURFACE TRACING RESULTS

7.1 Specimen Macro-Geometry

Figures 10 and 11 show a succession of 7 traces made on an unrun ground specimen as a preliminary appraisal of specimen macro-geometry. The trace made in a direction of 90° to the marked face of the specimen exhibits an error in form characteristic of the scuffing test specimen. If the contour had had the nominal crown radius the trace would appear as a straight line with fluctuations due to surface roughness alone. The deep valley and hill represent a departure from the ideal crowning radius. A consequence of this departure is that a low magnification level must be used in order to stay within the range of the The result is that the roughness Talysurf output limits. itself spans a relatively narrow range in the output signal. Whenever the roughness is a small percentage of full scale, the magnitude of quantization error is increased. Unfortunately, there is no convenient way in which this can be avoided.

Another unwanted effect of the non-ideal macro-geometry is that the macro-stress distribution across the disks is affected, most likely in a uniquely different way for each specimen pair, adding to overall test variability.

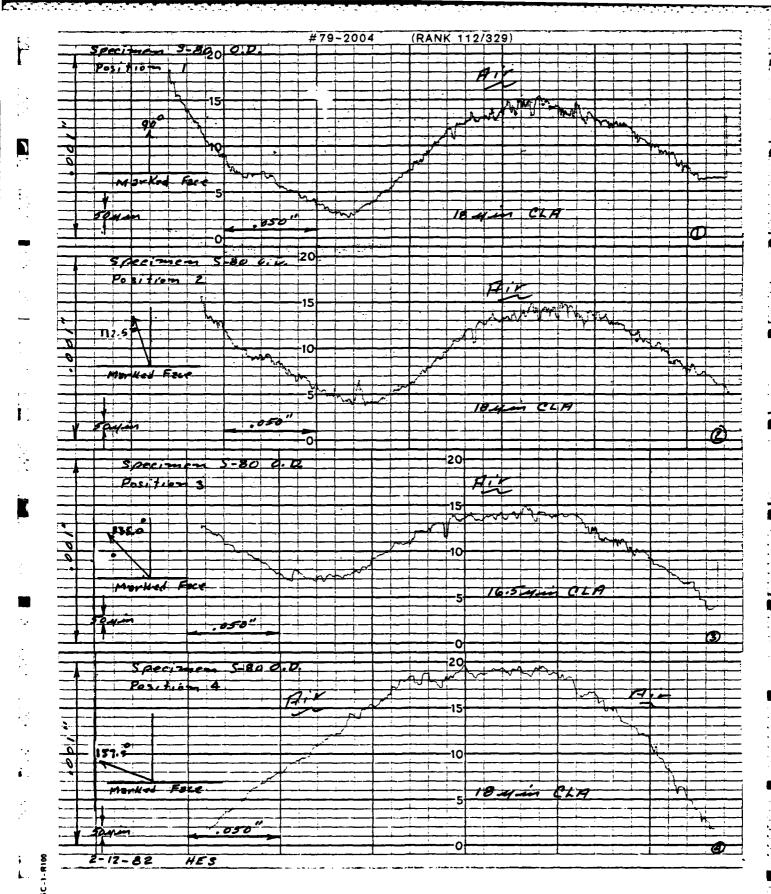


FIGURE 10; SPECIMEN MACRO-GEOMETRY VARIOUS TRACING ANGLES

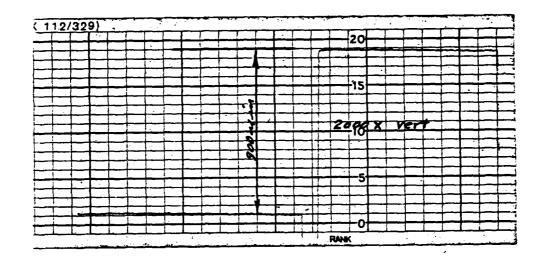
FIGURE 11: SPECIMEN MACRO-GEOMETRY VARIOUS TRACING ANGLES (CONTINUED)

The two traces made at angles $22\frac{1}{2}^{\circ}$ to the circumferential direction are close enough to being circumferential that the hill and valley macro-geometry is not seen.

The traces discussed above indicated that vertical magnifications of 2,000X or 5,000X would be necessary. Figure 12 shows traces over two calibration steps selected so as to occupy most of the dynamic range of the Talysurf at vertical magnifications of 2,000X and 5,000X. The respective step heights are 900 and 300 micro-inches.

7.2 Acquisition of Digitized Roughness Data

Tables 2 and 3 show the measured values of m₀, m₂, and m₄ obtained from PRODOE for 15 specimens. The first column in these tables is a specimen serial number assigned by NAPC along with a test number as specified in the scuffing test summary in Table 1. The letter F or S is prefixed to the test number to indicate whether this specimen was the fast or slow member of the mating pair. The second column shows the tracing angle with 0° representing the axial direction on the specimen and 90° the circumferential direction. As noted previously, however, the 90° direction was omitted in the measurement work. As noted in Appendix II, the value of m₄ is constrained by reason of the stylus tip radius to be less than 10^{-4} (μ in)⁻². There are several



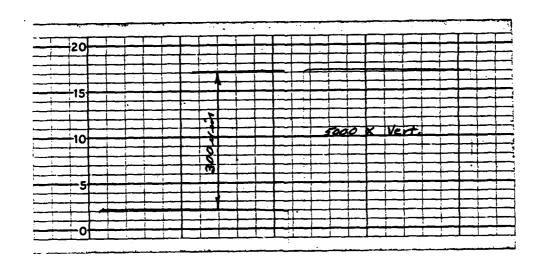


FIGURE 12: CALIBRATION STEPS

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SURFACE ROUGHNESS DATA -	7 ANGULARLY SPACED PROFIL	PER SPECIMEN
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TABLE		

SPECTIMEN NO. / PTLE NO.	YEN	9	ZEI	Į	SPECIMEN NO./FILE NO.	TEN	OE	Z _{EE}	2
	(090)	(utn ²)		(vtn ⁻²)		(DBC)	(µln²)	ļ	(pln ⁻²)
S-63/FIA (GROMD)	22.5 45.0 45.0 67.5 112.5 1135.0	5.69563x10 ² 5.78942x10 ² 5.72919x10 ² 5.57345x10 ² 5.69731x10 ² 5.83593x10 ² 5.74799x10 ²	8.91980x10 ⁻¹ 8.99846x10 ⁻¹ 8.98541x10 ⁻¹ 8.95642x10 ⁻¹ 8.9690x10 ⁻¹ 9.14146x10 ⁻¹ 8.87857x10 ⁻¹	3.01825x10 ⁻³ 2.9989x10 ⁻³ 3.04607x10 ⁻³ 3.03515x10 ⁻³ 3.01703x10 ⁻³ 2.96795x10 ⁻³	8-85/F4A (FOLISHED)	22.5 45.0 67.5 112.5 135.0 157.5	7.08460x10 ¹ 7.27218x10 ¹ 7.30595x10 ¹ 7.37921x10 ¹ 7.47322x10 ¹ 8.04307x10 ¹ 7.99206x10 ¹	1.07010×10 ⁻¹ 1.12950x10 ⁻¹ 1.14064x10 ⁻¹ 1.1269x10 ⁻¹ 1.1269x10 ⁻¹ 1.16964x10 ⁻¹ 1.19991x10 ⁻¹ 1.24397x10 ⁻¹	3.61782x10 ⁻⁴ 3.7225x10 ⁻⁴ 3.90229x10 ⁻⁴ 3.72361x10 ⁻⁴ 3.92476x10 ⁻⁴ 3.9639x10 ⁻⁴ 4.17527x10 ⁻⁴
8-64/SIA (GRUND)	22.5 45.0 67.5 112.5 113.0	5.69465x10 ² 5.49601x10 ² 5.77393x10 ² 5.62104x10 ² 5.6371x10 ² 6.08779x10 ² 5.73581x10 ²	8.72614x10 ⁻¹ 9.63145x10 ⁻¹ 9.10018x10 ⁻¹ 9.18199x10 ⁻¹ 8.6423x10 ⁻¹ 9.3720x10 ⁻¹ 8.90843x10 ⁻¹	2.90730k10 ⁻³ 2.92024k10 ⁻³ 3.02527k10 ⁻³ 3.01544k10 ⁻³ 2.92810k10 ⁻³ 3.09104k10 ⁻³ 3.00566k10 ⁻³	8-74/54A (POLISHED)	22.5 45.0 67.5 67.5 112.5 135.0	8.68919x10 ¹ 6.92640x10 ¹ 8.88909x10 ¹ 8.79658x10 ¹ 8.85773x10 ¹ 9.33314x10 ¹	1.30094x10 ⁻¹ 1.36137x10 ⁻¹ 1.37634x10 ⁻¹ 1.40362x10 ⁻¹ 1.39826x10 ⁻¹ 1.40395x10 ⁻¹	4. 29721x10 ⁻⁴ 4. 53442x10 ⁻⁴ 4. 65841x10 ⁻⁴ 4. 70870x10 ⁻⁴ 4. 6346x10 ⁻⁴ 4. 63460x10 ⁻⁴
S-3/F3A (GROUND)	22.5 45.0 45.0 67.5 112.5 113.0 157.5	5.06756x102 4.97629x102 5.91970x102 5.38903x102 5.28337x102 5.26889x102 5.95163x102	7.78856x10 ⁻¹ 7.7526x10 ⁻¹ 8.09345x10 ⁻¹ 8.18245x10 ⁻¹ 8.21256x10 ⁻¹ 9.23453x10 ⁻² 9.18786x10 ⁻¹	2.5991x10 ⁻³ 2.5591x10 ⁻³ 2.6729x10 ⁻³ 2.6729x10 ⁻³ 2.7312x10 ⁻³ 2.7312x10 ⁻³ 2.7356x10 ⁻³ 3.04950x10 ⁻³	8-13/FSA (dicump)	22.5 45.0 67.5 112.5 135.0	5.61149x102 6.02475x102 5.79291x102 5.86173x102 5.85069x102 5.8725x102 6.05785x102	8.73582x10 ⁻¹ 8.83669x10 ⁻¹ 8.90960x1 ⁻¹ 9.03269x10 ⁻¹ 8.97964x10 ⁻¹ 9.19718x10 ⁻¹	2.91865x10 ⁻³ 2.33140x10 ⁻³ 2.99269x10 ⁻³ 3.00972x10 ⁻³ 3.06100x10 ⁻³ 3.28697x10 ⁻³
8-4/SJA (GROUND)	0 22.5 45.0 67.5 112.5 135.0	9.21712x10 ¹ 9.04377x10 ¹ 9.06914x10 ¹ 8.89149x10 ¹ 9.22148x10 ¹ 9.28104x10 ¹	1.43670x10 ⁻¹ 1.41903x10 ⁻¹ 1.38120x10 ⁻¹ 1.38077x10 ⁻¹ 1.43408x10 ⁻¹ 1.42657x10 ⁻¹	4.75339x10* 4.69920x10* 4.67246x10* 4.60750x10* 4.63759x10* 4.1375x10*					

SPECIFIEM NO./FILE NO.	ANG.CE (DBG)	^m 0 (utn ²)	Z _M	mq.,(utn-2)	SPECIMEN NO./FILE NO.	ANGLE (DBG)	m ₀ (uin ²)	Z _{III}	74 (vin ⁻²)
-11/FGA (FOLISED)	0 22.5 45.0 67.5 112.5 113.0	9.4133#10 ¹ 9.1748#10 ¹ 9.69677#10 ¹ 9.5231#10 ¹ 9.33078#10 ¹ 9.33076#10 ¹	1.47655x10 ⁻¹ 1.4345xx10 ⁻¹ 1.47545xx10 ⁻¹ 1.4041xx10 ⁻¹ 1.42017x10 ⁻¹ 1.46267x10 ⁻¹ 1.4257xx10 ⁻¹	4.93298x10 ⁻⁴ 4.89300x10 ⁻⁴ 4.62518x10 ⁻⁴ 4.78664x10 ⁻⁴ 4.89693x10 ⁻⁴ 4.79344x10 ⁻⁴	S-15/F58 (GROCIAD)	0 22.5 45.0 67.5 112.5 135.0	6.17272x10 ² 5.91661x10 ² 4.1678x10 ² 1.15102x10 ³ 1.50233x10 ³ 3.71844x10 ² 4.34670x10 ²	3.02643x10 ⁻² 2.61151x10 ⁻² 9.11312x10 ⁻³ 2.59011x10 ⁻² 3.5385x10 ⁻² 1.57065x10 ⁻² 3.10046x10 ⁻²	4.78554x10 ⁻⁵ 9.13416x10 ⁻⁶ 9.13416x10 ⁻⁶ 7.30478x10 ⁻⁵ 3.22097x10 ⁻⁵ 6.64262x10 ⁻⁵
5-12/96A (POLISHED)	22.5 45.0 67.5 112.5 135.0	5.5649x102 5.32763x102 5.53767x102 5.61536x102 5.47514x102 5.48620x102 5.5544x102	8. 67324x10 ⁻¹ 8. 25456x10 ⁻¹ 8. 68822x10 ⁻¹ 8. 62498x10 ⁻¹ 8. 44894x10 ⁻¹ 8. 50930x10 ⁻¹	2. 87862x10 ⁻³ 2. 74242x10 ⁻³ 2. 91568x10 ⁻³ 2. 88005x10 ⁻³ 2. 84525x10 ⁻³ 2. 87200x10 ⁻³ 2. 85439x10 ⁻³	8-16/858 (GROUND)	0 22.5 45.0 67.5 112.5 113.0	2.34027x102 3.30695x102 5.4407x102 4.58701x103 2.74058x103 6.47675x102	1.73835x10 ⁻² 1.58166x10 ⁻² 6.36499x10 ⁻² 4.37791x10 ⁻² 2.62259x10 ⁻² 1.66626x10 ⁻²	4.65714x10 ⁻⁵ 4.04731x10 ⁻⁵ 4.04731x10 ⁻⁵ 9.38172x10 ⁻⁵ 7.05630x10 ⁻⁵ 5.91834x10 ⁻⁵ 3.67364x10 ⁻⁵
9-9/F48 (POLISHED)	22.5 45.0 67.5 112.5 1135.0	3.66328x10 ² 1.54228x10 ² 7.37750x10 ¹ 6.67592x10 ² 7.14054x10 ² 7.66339x10 ² 3.59623x10 ²	4.35498x10 ⁻³ 3.54741x10 ⁻³ 3.43898x10 ⁻³ 9.2263x10 ⁻³ 3.41394x10 ⁻³ 4.05602x10 ⁻³	8.23214x10 ⁻⁶ 7.28907x10 ⁻⁶ 7.6528x10 ⁻⁵ 1.31735x10 ⁻⁵ 1.5290x10 ⁻⁵ 7.32113x10 ⁻⁶ 7.61359x10 ⁻⁶	8-83/FGB (FOLLISHED)	0 22.5 45.0 67.5 113.0 157.5	1.03611x102 1.77932x102 1.02237x102 6.07290x102 6.89468x102 1.59673x102	1.10226x10 ⁻² 8.69739x10 ⁻³ 5.93537x10 ⁻³ 1.18710x10 ⁻² 1.18971x10 ⁻³ 1.08477x10 ⁻²	1.64349x10 ⁻⁵ 1.40258x10 ⁻⁵ 1.3506x10 ⁻⁵ 1.97382x10 ⁻⁵ 2.3225x10 ⁻⁵ 1.90516x10 ⁻⁵ 2.03973x10 ⁻⁵
F10/948 (PCL194ED)	27.5 45.0 67.5 112.5 113.6 157.5	1.67963x10 ² 3.6963x10 ² 5.99646x10 ² 7.69133x10 ² 1.69731x10 ²	3.51377x10 ⁻³ 3.674x10 ⁻³ 3.674x10 ⁻³ 1.01124x10 ⁻³ 9.01035x10 ⁻³ 9.01035x10 ⁻³ 3.69695x10 ⁻³	7. 221 22x10-6 6. 6. 8158n10-6 6. 0224 x10-6 1. 50680x10-5 1. 3505x10-5 8. 79753x10-6 8. 79753x10-6	9-8/968 (POL.134ED)	0 6.831 • 22.5 45.0 6.839 67.5 2.773 112.5 1.786 137.5 1.653	6.83183x10 ² 6.83931x10 ² 2.2736x10 ³ 1.8689x10 ³ 1.65369x10 ² 1.65369x10 ² 2.878CR	3.20475x10 ⁻² 1.78795x10 ⁻² 5.8252x10 ⁻² 5.8252x10 ⁻² 2.17155x10 ⁻² 2.02902x10 ⁻²	8.4860x10 ⁻⁵ 5.76405x10 ⁻⁵ 1.2641x10 ⁻⁴ 7.62121x10 ⁻⁵ 6.22460x10 ⁻⁵

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instances in the data of Tables 2 and 3 in which the measured value of m_{μ} exceeds 10^{-4} . This is taken to be evidence of extraneous, high frequency noise from electronic or vibration sources. A second observation is that the value of mo is very consistent, i.e. varies very little, with respect to tracing direction in several of the tests. Inspection reveals that the specimens for which mo is consistent coincides with the specimens for which m_{\parallel} exceeds 10^{-4} . One explanation could be that the extraneous, high frequency noise which causes mu to be too large is providing a high level background which tends to mask any true differences with tracing direction. For specimens which appear consistent, the value of m_0 is in the neighborhood of 500-600 (μ $in)^2$ for the ground specimens and in the neighborhood of 70 to 90 $(\mu \text{ in})^2$ for the polished specimens. Specimens S3A and S6A, however, are exactly the opposite to this observation raising the question of whether the data for these specimens might have been reversed. Examining the order in which the data were taken, there appeared to be a clear relationship between variability in mo with respect to tracing direction, and the order of data acquisition, suggesting a reduction in scatter as the operator acquired skill in the use of the newly developed system.

Because of these various anomalies that were noted in the data it was decided to repeat the entire data acquisition process

but confining the measurements to a single trace in the axial direction. It was further decided to:

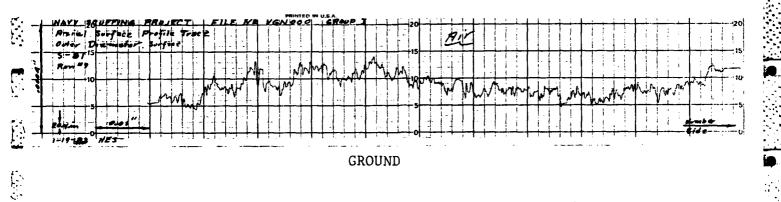
- 1. Reduce the number of data points acquired to 5,000, thereby both speeding the acquisition time and confining the entire trace to within the running track.
- 2. Change the upper filter setting as discussed in Appendix III to 2,750 cycles per inch. This value is calculated so as to assure that stylus radius effects are eliminated.
- 3. Make two traces in the unrun area of both a polished and ground specimen for reference purposes.

7.3 Repeated Tracing-Axial Direction

Figures 13 through 17 show sections of the Talysurf traces obtained in the repeat measurements. Figure 13 shows the traces over the unrun area of polished specimen No. S-77 and ground specimen No. S-81. All of the traces in Figures 13 through 14 were made at magnification of 5,000X and thus are directly comparable. Figure 13 confirms the difference in roughness in the polished and ground specimens and is consistent with the SEM photographs of the unrun sections of these two specimens shown in

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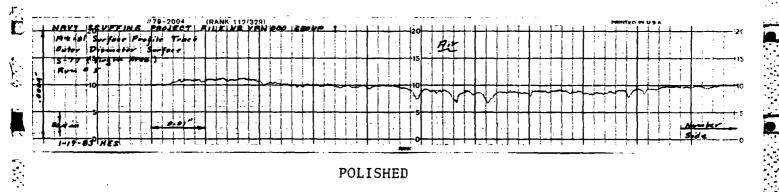


FIGURE 13: PROFILE TRACES OF UNRUN SURFACES

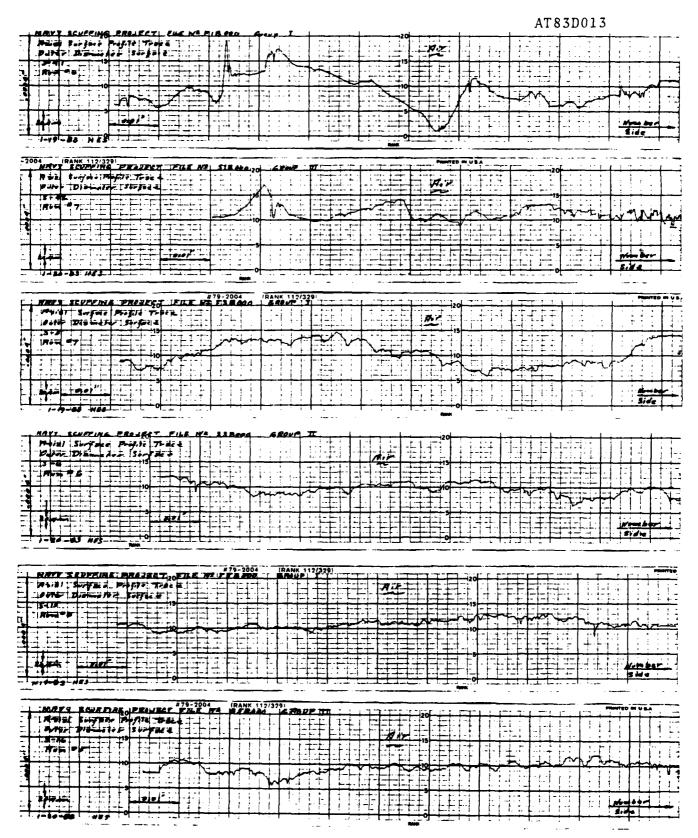


FIGURE 14: PROFILE TRACES OF RUN SURFACES
GROUND - UNSCUFFED

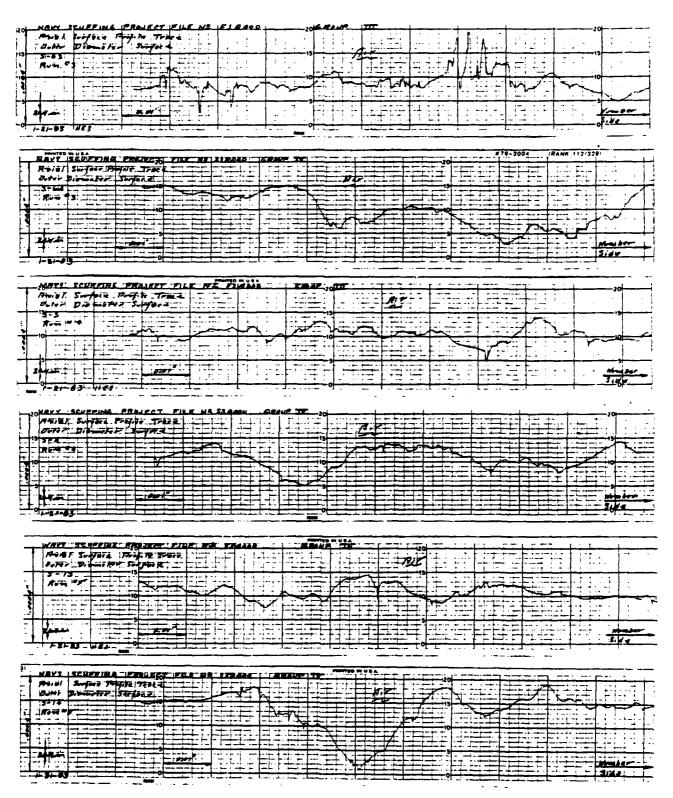


FIGURE 15: PROFILE TRACES OF RUN SURFACES

GROUND - SCUFFED

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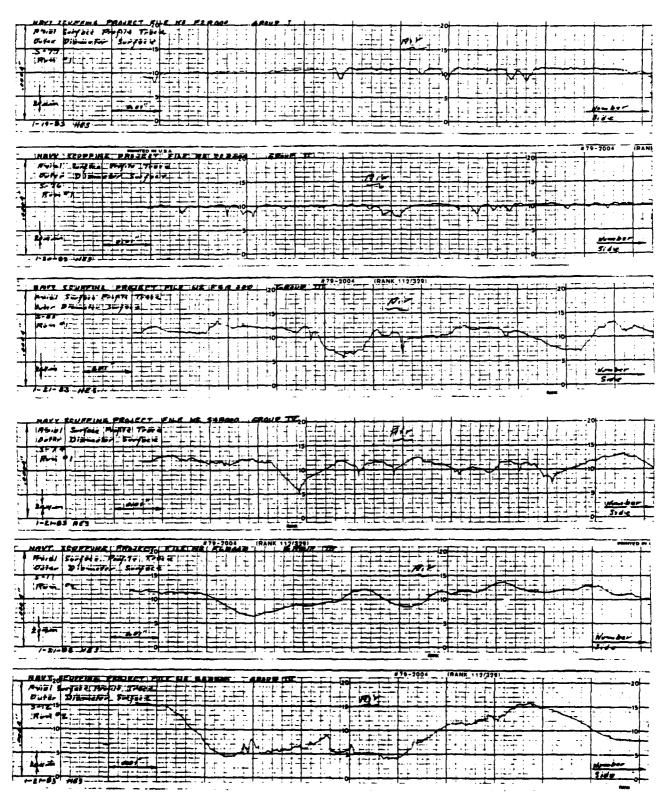
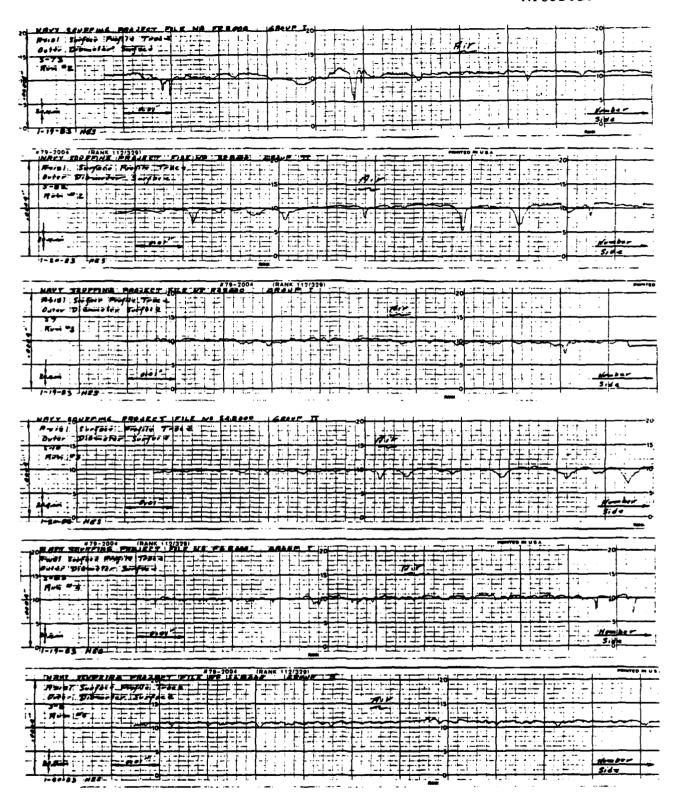


FIGURE 16: PROFILE TRACES OF RUN SURFACES POLISHED - SCUFFED



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FIGURE 17: PROFILE TRACES OF RUN SURFACES
POLISHED - UNSCUFFED

Figure 7. Figure 14 shows the ground specimens that scuffed;
Figure 15 the ground specimens from the tests that were not run to scuffing. Figure 16 shows the polished specimens that scuffed and Figure 17 the polished specimens that did not.

Qualitatively, the ground, scuffed specimens seem to be distinguishable from the polished, unscuffed, specimens, but in all, the differences among the four categories are surprisingly small. Within the ground specimens it is difficult to argue that the scuffed and the unscuffed are distinguishable. Within the polished specimens one could argue that the scuffed appear to be rougher than the unscuffed.

Figure 18 shows the histogram produced by PRODOE for an unrun portion of the ground surface of specimen No. S-81. It is reasonably symmetrical, suggestive of a gaussian height distribution. Figure 19 is the histogram of an unrun portion of polished specimen No. S-77. This histogram shows a longer left tail consistent with the fact that polishing tends to remove peaks while leaving valleys undisturbed. Histograms for the four specimen categories, "ground-scuffed", "ground-unscuffed", "polished-scuffed", "polished-unscuffed", are included in Appendices IV through VII for reference.

Table 4 shows the values of m_{0} , m_{2} , and m_{4} obtained from the single axial profiles with high-pass filtering at 33 cycles/in.

HISTOGRAM OF FILE DK VGNOOOFF

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NORMAL IZED	CUMULATIVE	0.0068	0.0272	0.0340	0.0612	0.1429	0.2109	0.3299	0.4660	0.5884	0.7517	0.8537	0.9184	0.9694	0.9898	0.9932	1.0000
UPPER CELL	BOUNDARY	-0.67226E+02				-0.29621E+02	-0.20219E+02	-0.10818E+02	-0.14166E+01	0.79847E+01	0.17386E+02	0.26787E+02	0.36189E+02	0.45590E+02			0.73794E+02

FIGURE 18: HISTOGRAM - UNRUN GROUND SURFACE

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TE FREQUENT 1 2 44 44 51 52 44 16 16 17 17 17	ACCOCOCOCOCOCOCOCOCOCOCOCOCOCOCOCOCOCOC	UPFER CELL BOUNDARY -0.35469E402 -0.31337E402 -0.23073E402 -0.18941E402 -0.18961E402 -0.18961E402 -0.18961E402 0.17186E401 0.58506E401 0.58506E401 0.99825E401	NORMAL IZED	CUMULATIVE FREQUENCY				0.0170 3	0.0238	0.0408	0.0612 6	0.1156 16		0.3878	0.5612	0.7075	0.8333	0.9320	0.9796 1.4	
---	---	--	-------------	-------------------------	--	--	--	----------	--------	--------	----------	-----------	--	--------	--------	--------	--------	--------	------------	--

- UNRUN POLISHED SURFACE HISTOGRAM FIGURE 19:

TOTAL NUMBER OF FOINTS

TABLE 4: SPECTRAL MOMENTS FROM SINGLE AXIAL TRACE AFTER FILTERING - UPPER FILTER SETTING = 2750 CYCLES/IN.

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and low-pass filtering at 2,750 cycles/in. The values are grouped into four categories: "ground-scuffed", "ground-unscuffed", "polished-scuffed", and "polished-unscuffed". There is substantial variability in m_0 , m_2 , and m_4 within each of the four categories. Also shown are the corresponding values for the unrun ground and polished surfaces.

7.4 Variability

Church [13,21] has derived theoretical expressions for the coefficient of variation (standard deviation divided by the mean) of spectral moments corresponding to spectra of the form

for various n. Sayles & Thomas [14] have compiled numerous examples showing the universality of the "inverse square" spectrum corresponding to the special case of the above spectrum having n = 2. For this case Church's expression for the coefficient of variation of m_0 is

$$\gamma_0 = (1/\sqrt{3}) (\sqrt{D}/L)$$

where D is the period of the lowest frequency present in the data and L is the record length. With $\Delta X = 14.5E-6$ in. and N = 5,000 points, the record length L is

$$L = N\Delta X = 5,000 \bullet 14.5E-6 = 0.07236 in.$$

By filtering at a lower cutoff frequency of f=33 cpi, the value of the lowest frequency present is D=1/33=0.0303 in. Thus, the coefficient of variation of m_0 is predicted to be

$$\gamma_0 = 1/\sqrt{3} \sqrt{0.0303}/0.07236 = 0.374$$

The observed coefficient of variation of m_0 in each of the four categories shown on Table 4 are:

			Polished-Unscuffed
0.550	1.21	0.590	0.885

The fact that the observed values are higher than theoretical suggests that there are real differences among the specimens in each category, i.e. the spread is larger than expected due to chance alone. Nonetheless, the statistical variability is quite high, because D is large relative to L.

8.0 DATA ANALYSIS

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The values of m_0 , m_2 and m_4 from Table 4 are repeated in Table 5. These values were used as input to program ASPERSIM to compute the micro-contact conditions corresponding to the contact of the slow and fast disks of each test. The disk surfaces were regarded as isotropic and separated by a distance equal to the calculated plateau film thickness calculated at the conjunction temperature. Based on the spectral moments m_0 , m_2 at m_4 from a single trace on an isotropic surface the 9 bispectral moments needed for ASPERSIM input are:

$$m_{02} = m_{20} = m_2$$

$$m_{11} = m_{13} = m_{31} = 0$$

$$m_{22} = m_{4/3}$$

$$m_{O4} = m_{4O} = m_4$$

$$m_{OO} = m_{O}$$

Table 5 lists for the four specimen categories the ASPERSIM output:

ZNCON: Contacts per in²

TOTF: Elastically supported load per unit

AVF: The area average force supported by individual

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asperity contacts

OUTPUT
AND ASPERSIM
AND
MOMENTS
SPECTRAL MOMENTS
5:
TABLE

	SPEC.	TEST	•	Jc-31	•	2	2	-	N/6		a	MO. UF	CONTACTS	CONTRACT	ACPEDITY	TOTA COMTACT	AVA	
	į	į	-	CONJ.	_			(ntn-2)					PER In?	LOAD/UNIT	_	AREA/	ASPERITY	
			(1bs)		(rfa)	(utn2)		×10-7		(at.)	£	£		AREA		UNIT AREA	CONTACT AREA	_
														(1bs/1n ²)	(Jbs)			_
GROUND -	\$-63	F13	996	457	47.9	1150	9.07E-3	7.61	0.082	9.15E-2	2.13E-2	1.12754	1.8456	4.1755	0.227	0.223	1.216-7	
SCUFFED	2-64	SIA	996	457	47.9	1143	1.26E-3	1.57	0.082	9.15E-2	2.13E-2	1.12764	1.84E6	4.17.5	0.227	0.223	1.216-7	
	S-3	F38	3516	415	37.7	969	8.64E-4	918	0.099	14.1 E-2	3.286-2	2.70254	1.86E6	3.9365	0.211	0.392	2.10E-7	_
	7-0	S3A	3516	415	37.7	256	5.596-4	.654	0.099	14.1 E-2	3.286-2	2.70264	1.86E6	3.9355	0.211	0.392	2.10£-7	_
	5-13	FSA	1466	367	52.5	574	9.61E-4	1.13	0.141	10.5 E-2	2.44E-2	1.553E4	1.93E6	5.5985	0.289	0.403	2.08E-7	_
	\$-14	SSA	1466	367	52.5	2185	1.55E-3	1.87	0.141	10.5 E-2	2.44E-2	1.553£4	1.93E6	5.5985	0.289	0.403	2.08E-7	
GROUND -		F18	483	341	52.3	1908	3.68E-3	3.93	0.112	7.26E-2	1.69E-2	7.401E3	1.9266	4.9955	0.260	0.299	1.568-7	
UNSCUFFED		SIB	483	341	52.3	831	1.595-3	1.80	9.112	7.26E-2	1.691-2	7.401E3	1.9266	4.9985	0.260	0.299	1.56E-7	
	S-5	F.38	1758	392	21.1	248	1.48E-3	2.28	0.201	11.2 E-2	2.606-2	2.388E4	2.61E6	2.025	0.01	0.217	8.296-8	
	S-6	85 85	1758	392	21.12	961	8.236-4	.973	0.201	11.2 E-2	2.605-2	2.388E4	2.6166	2.02E5	0.077	0.217	8.295-8	
	S-15	F58	718	98	17.0	126	1.22E-3	1.69	0.588	8.33E-2	1.946-2	1.000E4	1.9756	9.1464	0.046	0.116	5.92E-8	
	S-16	828	718	906	17.0	791	9.68E-4	1.24	0.588	8.33£-2	1.945-2	1.000£4	1.97E6	9.1464	0.046	0.116	5.92E-8	_
																		-
POL I SHED -	5-77	FZA	9911	425	16.4	142	6.96E-4	.821	0.260	9.74E-2	2.27E-2	1.40354	2.02E6	1.2965	0.064	0.189	9.376-8	_
SCUFFED	S-76	SZA	1166	425	16.4	821	6.73E-4	.673	0.260	9.746-2	2.27E-2	1.40354	2.02E6	1.29£5	0.064	0.189	9.376-8	_
	S-85	F4A	3716	425	33.7	433	9.05E-4	.924	0.226	14.4 E-2	3.346-2	2.659E4	1.7656	2.7955	0.158	0.271	1.546-7	
	S-74	SAA	3716	425	33.7	<u>1</u> 02	1.096-3	1.27	0.226	14.4 E-2	3.346-2	2.659E4	1.7666	2.79£5	0.158	0.271	1.546-7	_
	S-11	F6A	2116	164	9.62	342	2.47E-4	.335	0.177	11.9 E-2	2.76E-2	2.218E4	2,15E6	2.9265	0.136	0.302	1.416-7	_
	S-12	X6A	2116	16+	9.62	536	1.40E-3	1.78	0.177	11.9 E-2	2.76E-2	2.218E4	2.15E6	2.92E5	0.136	0.305	1.416-7	
					1													_
POLISHED -	S-73	F28	583	312	23.4	295	2.68E-3	5.69	0.276	7.73E-2	1.805-2	7.868E3	1.8056	1.7165	0.095	0.167	9.286-8	_
UNSCUFFED	S-81	82S	583	312	23.4	522	1.03E-3	.633	0.276	7.73E-2	1.805-2	7.868E3	1.8056	1.7165	0.095	0.167	9.285-8	
	8-9	1.48	1858	320	12.2	53	6.90E-4	1.0	0.398	11.4 E-2	2.65E-2	2.325E4	2.45E6	8.9954	0.037	0.149	6. tot-8	_
	S-10	95	1858	320	12.2	16	5.50E-4	.723	0.398	11.4 E-2	2.65E-2	2.325E4	2.45£6	8.9954	0.037	0.149	6.10E-8	_
	5-83	F68	1058	326	8.83	ß	1.426-3	2.27	1.01	9.446-2	2.196-2	1.34464	2.0766	3.0954	0.015	0.056	2.71E-8	_
	8-8	898	1058	326	8.8	\$2	3.785-4	.427	1.01	9.44E-2	2.19E-2	1.344E4	2.0756	3.09£4	0.015	0.056	2.71E-8	_
		_	_	_				_										_

TOTA:

The elastic contact area per unit nominal area i.e.

the ratio of real to apparent contact area

AVA:

The area of an average asperity contact

The loads and areas computed by ASPERSIM are based on the assumption that the micro-contacts deform elastically. Even though with thin films, most if not all, asperities will undergo plastic deformation, the ASPERSIM output is descriptive of the severity of surface conditions, reflecting as it does the conditions that prevail prior to the onset of plastic flow.

Also listed in Table 5 are the load at scuffing or test termination, calculated conjunction temperature, the standard deviation of the composite surface of the two mating disks $\sigma = [(m_0) \text{fast} + (m_0) \text{slow}]^{1/2}, \text{ the contact ellipse major and minor axes a and b based on the scuffing load for the scuffed specimens or on the load at test termination for the unscuffed specimens, and n, the expected number of asperity contacts over the macrocontact ellipse computed as the product of ZNCON and the area of the contact ellipse mab.$

8.1 Conjunction Temperature

Figure 20 is a plot of scuffing load against conjunction temperature. The plot illustrates that the conjunction temperature is not constant at scuffing for either ground or polished specimens, nor is there a clear relationship between scuffing load and conjunction temperature.

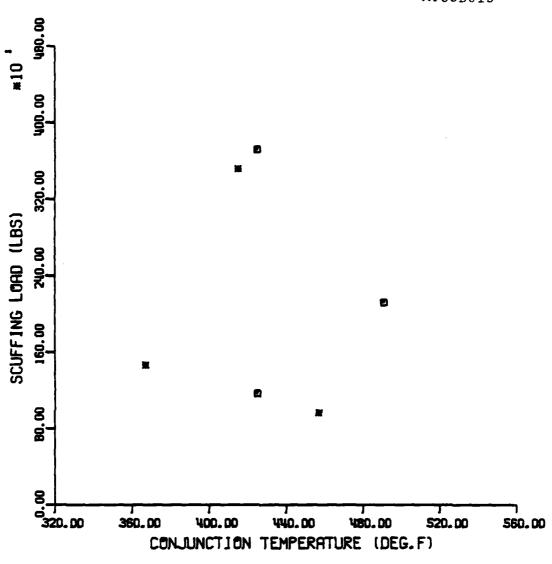
8.2 Film Parameter

Figure 21 is a plot of scuffing load against film parameter h/σ . For constant test conditions one might expect scuffing load to either increase with or at least not decrease with film parameter. This plot shows that the other factors, i.e. sliding speed, rolling speed and finish type are sufficiently influential to mask any film parameter effect which might exist.

8.3 Correlation of Scuffing Load With ASPERSIM Parameters

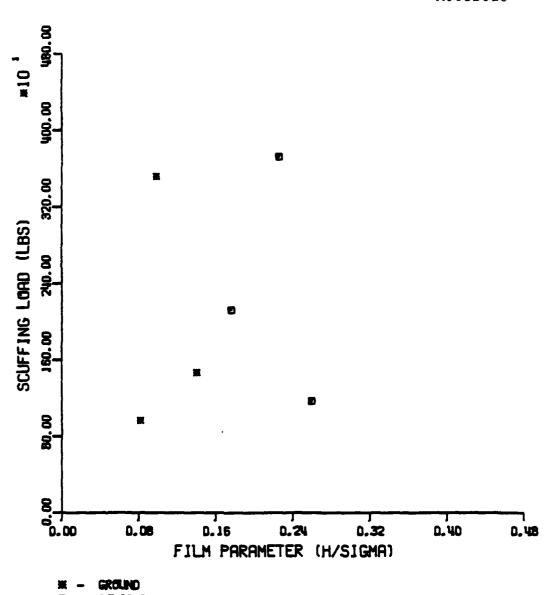
For the tests that proceeded to scuffing it is seen from the data in Table 5 that both n and AVA increase monotonically with scuffing load. AVF also increases monotonically with scuffing load for the polished specimen tests but fails to do so for the ground specimen tests. The fact that n increases with scuffing load is almost unavoidable since the contact area increases with load and ZNCON does not vary appreciably. Figure 22 is a plot of

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₩ - GROUND□ - POLISHED

FIGURE 20: SCUFFING LOAD
VS.
CONJUNCTION TEMPERATURE



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FIGURE 21: SCUFFING LOAD VS. FILM PARAMETER

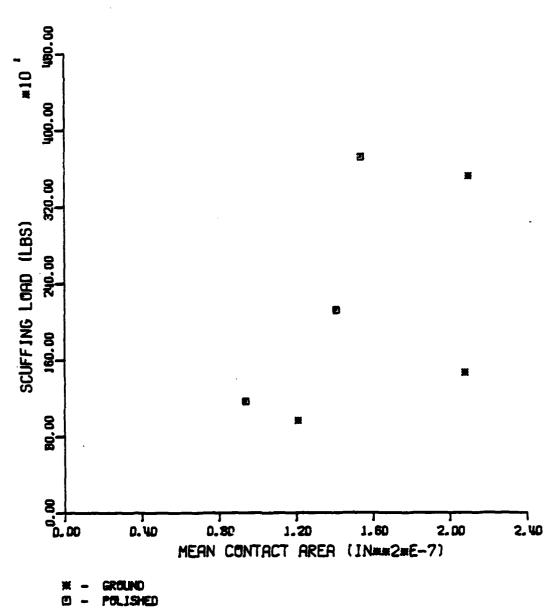


FIGURE 22: SCUFFING LOAD VS.
MEAN AREA OF CONTACT

scuffing load against AVA. The fact that scuffing load increases with AVA does not necessarily imply that large AVA values serve to increase the load at which scuffing occurs. More likely it reflects the evolutionary change in microgeometry with load. This is supported by the fact that for each kinematic condition AVA and AVF are higher for the ground specimens than for the polished specimens even though the scuffing load is always lower for the ground specimens. Run-in is discussed further in Section 8.5 below.

8.4 Comparison of Polished vs. Ground and Scuffed vs.

<u>Unscuffed Results</u>

Table 6 summarizes by finish type and scuffing status the average values of m_0 , m_2 and m_4 , the ASPERSIM output values, AVA, AVF and ZNCON, and the conjunction temperature Tc and the film parameter h/σ .

Program No. BMDP2V of the Biomedical System of Computer Programs for Statistical Analysis, 1 was used to evaluate the statistical significance of the observed differences in m_0 , m_2 and

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¹Winer [5] has, however, proposed a flash temperature correction factor of $[1-0.7\sigma]^{-1}$ (σ = CLA roughness in μ m).

TABLE 6: MEAN VALUES BY FINISH TYPE AND SCUFFING STATUS

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my between finish type and scuffing condition. In performing this analysis the 6 specimens at each combination of finish type (F) and scuffing condition (S) were treated as independent replicate tests. Actually since these six specimens were run at three different speed combinations they are not true replicates. The scatter among these tests is thus higher than it would be among pure replicates and the differences found to be significant by treating the data in this way are sure to be significant if pure replicates were used.

The analysis of variance tables are listed in Table 7. Effects for which the value listed under "Tail Probability" is less than 0.010 are statistically significant at the 10% level. With this criterion it is seen that m_0 differs significantly between the two finishes and between the scuffed and unscuffed specimens. The interaction is negligible however, meaning that the effect of finish is comparable for the scuffed and unscuffed specimens. m_2 and m_4 do not differ with respect to finish type or scuffing status at a 10% level of significance, but the m_4 effect is nearly significant between finish type.

Program BMDP2V was also applied to evaluate the significance of the observed differences between AVA, AVF and N. Since these quantities are determined by a pair of disks, there are only 3

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\$ 0.45617 1 65 3.24791 1 1 FRR17 8	0.6.5617 5.29.791 7.11110	0.00 U 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	0.772 0.518
ANALYSIS OF VARIANCE FOR 1-ST REPERPENT VARIABLE - 4	**************************************		
SOURCE SOURCE SOUNDE DESKEES	ES OF MEAN	L	TAIL PROBAPILITY
MEAN 129.15783 1	128.13783	26.23	0.001
S 000105 1	0.0105	0.00	0.989
38.39050			

TABLE 7: ANALYSIS OF VARIANCE SUMMARY - mo, m2 AND m4

observations for each combination of finish type and scuffing status. The analysis of variance summaries are given in Table 8. AVA and AVF both differ significantly with finish type and scuffing status, with ground specimens having larger average area and larger force.

The contact density is the same for scuffed specimens irrespective of original finish type. The scuffed specimens tend to have a lower contact density than the unscuffed specimens though the difference is not statistically significant.

Finally, BMDP2V was applied to the conjunction temperature values (Tc) and the film parameter values (HSIG). The respective analysis of variance summaries are given in Table 9. It is seen that Tc varies significantly with scuffing status but not at all with finish type. Of course, conjunction temperature must be higher for the scuffed specimens because flash temperature is a function of load and the unscuffed tests only reached half the load of the corresponding unscuffed tests.

For the film parameter there is a significant effect of scuffing status and an indication of an effect of finish. On balance, AVA and AVF are more sensitive than film parameter and, unlike conjunction temperature, indicate a finish effect.

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ANALYSIS OF VARIANCE SUMMARY - AVERAGE ASPERITY AREA (AVA), AVERAGE ASPERITY FORCE (AVF), AND AVERAGE DENSITY OF CONTACTS (ZNCON) .. ∞ TABLE

	SOMEE	SUM OF SQUARES	DEGREES OF FREEDOM	SOUARE
	اد المراباً ا	1759700.00000	1 176	8700.00000 216.74509
	\$ E	25483.93213 1950.74809	2	25483.39219 1350.74609
-	F443F	11439,39063		1429 - 92383
			;	
NALY	ANALYSIS OF VANIANCE FOR 1-ST	1-ST DEPENDENT VARIABLE - USIG	5156 - 316	
		SUM OF SQUARES	DEGREES OF FREEDOM	MEAN
	AF A V	1.05207		1.06207
	S	0.21335	1	0.01628
,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	ERADR	0.44281	&	0.05535
		•		
	TABLE 9: ANAL	ANALYSIS OF VARIANCE SUMMARY	YCF SHMMARY	ı

FATL PROBAFILITY

0.302 0.205 0.085 0.602

19.19 1.90 3.85 0.23

ANALYSIS OF VARIANCE SUMMARY -CONJUNCTION TEMPERATURE (TC) AND FILM PARAMETER (HSIG) IABLE

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PROBAL ILITY

ANALESIS OF VARIANCE FOR 1-ST DEPENDENT VARIABLE - TC

0.000 0.707 0.003 0.276

1236.92

8.5 Run-In

Run-in effects were not specifically studied in this investigation i.e. roughness characterizations were not made of the same surfaces at various stages of running. Nevertheless comparable surfaces on different specimens were characterized in the unrun condition, at the time of scuffing and at a half the time to scuffing.

Using the measured microgeometry of the unrun surfaces in Table 4, the surface velocities common to Tests 1A and 2A and the oil temperature (180°F) at start up, it was possible to compute the ASPERSIM variables for two ground and two polished disc specimens at start up conditions. The resultant values along with the values at the same velocities and at the conjunction temperature at half scuffing (Tests 1B and 2B) at scuffing (Tests 1A and 2A), are listed in Table 10.

In the same manner the microgeometry at half scuffing and at scuffing for tests 1A, 2A, 1B and 2B) was used with the 180°F start up oil temperature to calculate the comparable ASPERSIM characteristics of surfaces run-in to this extent. These values are shown in Table 10.

The parameter AVF is smaller for the polished than the ground surfaces and although AVF increases with running the difference

	h/o	ZNCON CONTACTS/ (in ²)	AVF (LBS)	AVA (in ²)	
GROUND	.367	3.23E6	.099	4.33E-8	(
POLISHED	.919	1.23E6	.033	5.77E-8	(UNRUN)
GROUND POLISHED	.268	1.75E6 1.40E6	.240	14.6E-8 7.82E-8	(HALF SCUFFING)
GROUND	.280	1.66E6	.203	11.1E-8	
POLISHED	.805	1.23E6	.044	7.14E-8	(AT SCUFFING)

TABLE 10: SURFACE PARAMETERS AT START UP CONDITIONS FOR THREE LEVELS OF RUN-IN

 $v_s = 138 \text{ IN/SEC}$

 $V_R = 345 \text{ IN/SEC}$

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between finishes persists. AVA on the other hand is initially smaller for the ground than the polished surfaces. For the run-in surfaces the polished specimens have a lower AVA value.

The contact density ZNCON, like AVF is also higher in each case for the ground than for the polished specimens at, each stage of run-in. This suggests the use of the product ZNCON x AVF, as a further candidate inverse measured of scuffing proneness. This parameter has the physical meaning of asperity supported force per unit area.

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APPENDIX I

SIGNAL PROCESSING FOR SURFACE ROUGHNESS PARAMETERS

APPENDIX I

SIGNAL PROCESSING FOR SURFACE ROUGHNESS PARAMETERS

Surface Roughness Characterization

The roughness or microscale topography of contacting bodies is generally recognized as a main determinant of their friction and wear behavior. The most commonly used measure of roughness of precision manufactured parts is the mean absolute deviation of the heights of a profile trace from their mean. This quantity, also known as the centerline average (CLA) or arithmetic average (AA) is the roughness measure adopted by the American National Standards Institute in Standard ANSI B46.1 on Surface Texture [16].

Other measures, dependent on the height distribution of profiles are in more limited use. One such is the root mean square (RMS) profile height; others are the skewness and kurtosis. RMS and CLA both measure essentially the same thing; scatter about the mean. For a given amplitude distribution function they are related to each other by a multiplicative constant e.g. for a gaussian amplitude height distribution function,

RMS = 1.25 • CLA. Skewness and kurtosis measure the asymmetry and peakedness of the distribution of profile heights and serve generally to detect departures of the height distribution function from the gaussian shape.

Two surfaces could differ drastically in the distance between, and in the sharpness of, the surface summits and still have the same CLA, RMS, skewness and kurtosis. Yet is is the density of summits and summit curvature as well as summit height that determines the severity and extent of the interaction that takes place when microscopically rough surfaces roll or slide on each other.

Moreover although surface <u>summits</u> determine the surface's performance the most commonly used equipment for surface roughness analysis is based upon stylus <u>profile</u> traces. Peaks on profiles generally represent the shoulders of asperities, i.e. the mean peak height will be lower than the mean summmit height. In addition, apparent frequencies on anisotropic surfaces will vary with the direction of tracing.

In [17] Nayak proposed a comprehensive system for characterizing general anisotropic surfaces as two dimensional random processes. Effectively this system overcame all of the short-comings of using just the CLA. This work specialized to solid surfaces the findings of Longuet-Higgins [18] for ocean waves.

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Two Dimensional Spectral Density

The point of departure in the two dimensional random process description is the notion of a two dimensional spectrum, whereby the total variability of the surface is decomposed into contributions from spatial frequencies measured in two orthogonal directions.

Let ω_{X} and ω_{Y} represent the radian frequency of the surface in the two orthogonal directions "x" and "y". The spectral density $S(\omega_{X}, \omega_{Y})$ has the property that the variability (variance) due to surface waves having frequencies between ω_{X} and $\omega_{X}+\Delta\omega_{X}$ in the "x" direction and between ω_{Y} and $\omega_{Y}+\Delta\omega_{Y}$ in the "y" direction is given for $\Delta\omega_{Y}$ and $\Delta\omega_{X}$ small, by

$$var[\omega_{x}, \omega_{y}] = S(\omega_{x}, \omega_{y}) \Delta \omega_{x} \Delta \omega_{y}$$
 (I-1)

The two dimensional spectral moment m_{ij} is defined as

$$m_{ij} = \iint \omega_{x} i \omega_{y} j S(\omega_{x}, \omega_{y}) d\omega_{x} d\omega_{y}$$
 (I-2)

Nayak showed that for a microhertzian model in which surface summits are approximated as elliptical paraboloids, the surface may be described by the even order two dimensional spectral moments of all orders up to four. This constitutes nine numbers: m_{00} , m_{02} , m_{11} , m_{20} , m_{04} , m_{13} , m_{22} , m_{31} and m_{40} .

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These nine values appear as parameters in the joint distribution of summit height and principal curvature and in the expressions for summit density.

Estimating Two Dimensional Spectral Moments

The link between profile traces and surface summit statistics is established by the relationship that these statistics have to the one dimensional spectral moments measured along a profile trace taken at an angle θ from the x direction.

Denoting frequencies measured in this direction by ω_θ and the associated (one dimensional) spectral density $S(\omega_\theta)$ then the even spectral moments of order 0, 2 and 4 are:

$$m_{O}(\theta) = \int \omega_{\theta}^{O} S(\omega_{\theta}) d\omega_{\theta}$$
 (I-3)

$$m_2(\theta) = \int \omega_{\theta}^2 S(\omega_{\theta}) d\omega_{\theta}$$
 (1-4)

$$m_4(\theta) = \int \omega_{\theta}^4 S(\omega_{\theta}) d\omega_{\theta}$$
 (I-5)

These values can be estimated in two ways, namely by either (1) estimating the spectral density and using numerical integration to approximate Eqs. (I-3) to (I-5) or (2) by using the relations:

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$$m_0 = E(z-\mu)^2 \tag{I-6}$$

$$m_2 = E(dz/dx)^2 (I-7)$$

$$m_4 = E[(dz^2/dx^2)^2]$$
 (I-8)

where z(x) denotes a profile made in direction x and μ is the mean line of the profile.

For digital processing given n profile height values $z_1...z_n$ equispaced a distance Δx apart, the quantities m_0 , m_2 and m_4 may be computed as:

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$$m_0 = \sum_{i=1}^{n} (z_{i-\mu})^2/n$$
 (I-9)

$$n-1$$
 $m_2 = \sum (z_{i+1} - z_i)^2/(n-1)(\Delta x)^2$
 $i=1$
(I-10)

$$m_4 = \sum_{i=2}^{n-2} (z_{i+1} - 2z_i + z_{i-1})^2 / (\Delta x)^4 (n-2)$$

$$i=2$$
(I-11)

Given the values of m_0 , m_2 and m_4 for traces made at different angles θ_1 , $\theta_2, \ldots, \theta_K$ on the surface with respect to an arbitrary coordinate system coincident with the surface mean plane, the bispectral moments can be fit by using multiple regression with the following theoretical relationships [cf. McCool [8]]

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$$^{m}O(\theta_{i}) = m_{OO}$$
 (I-12)

$$m_2(\theta_i) = m_{20}\cos^2\theta_i + 2m_{11}\cos\theta_i\sin\theta_i + m_{02}\sin^2\theta_i \quad (I-13)$$

$$m_{4}(\theta_{i}) = m_{40}\cos^{4}\theta_{i} + 4m_{31}\cos^{3}\theta_{i}\sin\theta_{i} + 6m_{22}\cos^{2}\theta_{i}\sin^{2}\theta_{i}$$
$$+ 4m_{13}\cos\theta_{i}\sin^{2}\theta_{i} + m_{04}\sin^{4}\theta_{i} \qquad (I-14)$$

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APPENDIX II

UPPER LIMIT ON m_4 IMPOSED BY STYLUS RADIUS

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APPENDIX II

UPPER LIMIT ON m4 IMPOSED BY STYLUS RADIUS

Church [19] has pointed out that the mean square curvature in stylus traces is limited by the stylus radius R.

A stylus of radius R cannot detect sharper curvatures than its own radius. Thus the measured curvature must exceed 1/R.

Let z(x) denote the equation of a profile. Since dz/dx is small, dz^2/dx^2 is approximately the curvature at any value of x and at the peaks where dz/dx = 0, it is exactly the curvature. Thus

$$1/d^2z/dx^2 > R \tag{II-1}$$

or
$$d^2z/dx^2 < 1/R$$
 (II-2)

and
$$(d^2z/dx^2)^2 < R^{-2}$$
 (II-3)

Since m_4 is the average value of $({\rm d}^2z/{\rm d}x^2)^2$ (cf. Appendix) then one must have

$$m_4 < R^{-2}$$
 (II-4)

For $R = 10^{-4} = 10^2 \mu$ in. the standard stylus radius used with the Talysurf IV, an upper limit on m_2 is

$$m_4 < (10^2)^{-2} = .0^{-4} (\mu in)^{-2}$$

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APPENDIX III

UPPER FREQUENCY DETERMINATION FOR BANDPASS FILTERING

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APPENDIX III

UPPER FREQUENCY DETERMINATION FOR BANDPASS FILTERING

A well known result [cf.20] is that a stylus of radius R will be able to touch the valleys of a sinusoidal surface of amplitude B so long as the frequency f does not exceed

$$f = 1/2\pi (RB)^{-1/2}$$
 (III-1)

The mean square height of a sinewave, m_0 , is

$$m_0 = B^2/2 \tag{III-2}$$

Therefore a random surface having mean square height $m_{\mbox{\scriptsize 0}}$ will have the same mean square height as an equivalent sinewave having amplitude

$$B = (2m_0)^{1/2}$$
 (III-3)

The upper frequency for traceability of the equivalent sinewave is then, from Eq.(III-1)

$$f = 1/2\pi (R(2m_0)^{1/2})^{-1/2}$$
 (III-4)

For $R = 10^{-4}$ in. this becomes

$$f = 13.38 (m_0)^{-1/4}$$

Using $m_0 = 575 \ (\mu in)^2$ for ground and $m_0 = 90 \ (\mu in)^2$ for polished specimens gives, after rounding to the nearest 50 cycles/in.

$$f = 2,750$$
(ground)

= 4,350 (polished)

It is quite likely that good information can be obtained without stylus distortion at frequencies higher than given by this calculation since the actual random surface will not have all of its energy concentrated at a single frequency as assumed in this calculation. It is therefore quite justifiable to assert that the signal is not affected by stylus distortion at frequencies below the limit calculated as above.

Inasmuch as the value of the upper cutoff has a strong effect on the number of contacts, the value 2750 has been used in processing both ground and polished specimen roughness data to assure comparability.

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APPENDIX IV

HISTOGRAMS

GROUND-SCUFFED SPECIMENS

F-4185C-1-R10

F-1185C-1-R100

TOTAL NUMBER OF POINTS

****** XXXXXXXXXXXXXXXXXXXXXXXXXX xxxxxxxxxxxxxxxxxx xxxxxxxxxxxxxxxx XXXXXXXXXXXXXXXXX xxxxxxxxxxxxxxxx XXXXXXXXXXXXXXXX xxxxxxxxxxxxx XXXXXXXXXXX ×××××××× XXXXXXX XXXXX FREQUENCY 13 16 24 26 2002222 10 0 CM FOINT NORMAL IZED CUMULATIVE FREQUENCY 0.0170 0.0374 0.0646 0.1088 0.1395 0.1939 0.2755 0.3639 0.4864 0.5850 0.6599 0.7959 0.8639 0.9320 0,7211 -0.10297E+03 -0.90661E+02 -0.78352E+02 -0.66043E+02 -0.53734E+02 -0.41425E+02 -0.29117E+02 -0.16808E+02 0.20118E+02 0.32427E+02 0.44736E+02 0.57045E+02 0.69354E+02 -0.44991E+01 0.78096E+01 UPPER CELL BOUNDARY

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0.81662E+02

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4000	0.0034	-0.87593E+02 -0.77048E+02
× + 0	0,0034	-0.98137E+02 -0.82593E+02
FREQUENCY	CUMULATIVE	BOUNDARY
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4	34	52	64	37	29	27	27	D ~	•0
 0.0306	0.1463	0.3231	0.5408	0.6667	0.7653	0.8571	0.9490	9626.0	1.0000
-0.3486BE+02	-0.24323E+02	-0.13778E+02	-0.32327E+01	0.73123E+01	0.17857E+02	0.28402E+02	0.38947E+02	0.49492E+02	0.60037E+02

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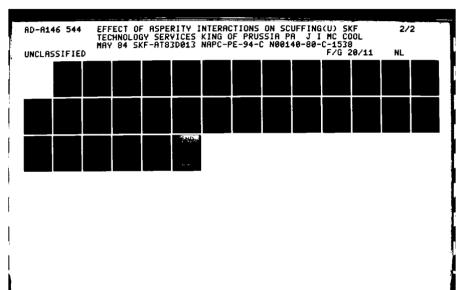
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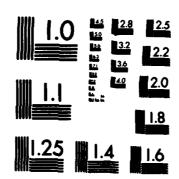
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TOTAL NUMBER

POINT FREQUENCY	0	6 XXXXXXXX	17 XXXXXXXXXXXXXXX	20 XXXXXXXXXXXXXXXXXX	6 XXXXXXXX	14 XXXXXXXXXXXX	33 XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX		42 XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX	30 XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX	24 XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX	24 XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX	12 XXXXXXXXXX	7 XXXXXXX	16 XXXXXXXXXXXXX	B XXXXXXX
NORMALIZED CUMULATIVE FREQUENCY	000000	0.0306	0.0884	0.1565	0.1871	0.2347	0.3469	0.4456	0.5884	0.6905	0.7721	0.8537	0.8946	0.9184	0.9728	1.0000
UPPER CELL BOUNDARY	-0.55485E+02	-0.47840E+02	-0.40194E+02	-0.32549E+02	-0.24903E+02	-0.17258E+02	-0.96127E+01	-0.19672E+01	0.56782E+01	0.13324E+02	0.20969E+02	0.28614E+02	0.36260E+02	0.43905E+02	0.51551E+02	0.59196E+02





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NORMALIZED CUMULATIVE FREQUENCY	0.0102	0.0238	0.0408	0.0646	0.1054	0.1803	0.2585	0.3571	0.5306	0.6837	0.7925	0.8605	0.9252	0.9932	9966.0	1.0000
UPPER CELL BOUNDARY	-0.70932E+02	-0.61901E+02	-0.52871E+02	-0.43840E+02	-0.34810E+02	-0.25779E+02	-0.16749E+02	-0.77182E+01	0.13123E+01	0.10343E+02	0.19373E+02	0.28404E+02	0.37434E+02	0.46465E+02	0.55495E+02	0.64526E+02
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POINT	FREQUENCY	9	٥	32	36	41	35	22	12	20	14	10	20	17	٥	7	~
NORMAL IZED	CUMULATIVE FREQUENCY	0.0204	0.0510	0.1701	0.2925	0.4320	0.5510	0.6259	0.6667	0.7347	0.7823	0.8163	0.8844	0.9422	0.9728	9966.0	1.0000
UPPER CELL	BOUNDARY	-0.50978E+02	-0.41737E+02	-0.32496E+02	-0.23255E+02	-0.14014E+02	-0.47727E+01	0.44684E+01	0.13710E+02	0.22951E+02	0.32192E+02	0.41433E+02	0.50674E+02	0.59915E+02	0.69156E+02	0.78397E+02	0.87638E+02

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F-4185C-1-R100

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294. OF POINTS TOTAL NUMBER

×××× FREGUENCY FOINT NORMAL I ZED CUMULATIVE FREQUENCY 0.0068 0.0102 0.0238 -0.10850E+03 -0.93934E+02 -0.79369E+02 UPPER CELL BOUNDARY

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> -0.64804E+02 -0.50239E+02 -0.35675E+02 -0.21110E+02

63 28 452

0.7585 0.9388 0.9728 .0000 1.0000 0.6054 0.9014 0.9864 0.22585E+02 0.51715E+02 0.66279E+02 0.37150E+02 0.80844E+02 0.95409E+02 0.10997E+03 0.80200E+01

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APPENDIX V

HISTOGRAMS

GROUND-UNSCUFFED SPECIMENS

	HISTOGRAM OF FILE	FILE DK F1B0000FF	0000FF	
	MAX = 0.206	0.20644E+03 MIN	= -0.93465E	465E+02
	TOTAL NUMBER	OF POINTS =		294.
	UPPER CELL	NORMALIZED	FOINT	
	BOUNDARY	CUMULATIVE FREQUENCY	FREQUENCY	NCY
	-0.74721E+02	0.0510	E.	XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX
	-0.55976E+02	0.1054	16	XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX
	-0.37232E+02	0.1565	15	XXXXXXXXXXXXXXX
	-0.1848E+02	0.2755	32	XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX
	0.25584E+00	0.5884	92	XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX
ν	0.19000E+02	0.7585	50	**************************************
- 2	0.37744E+02	0.8571	29	XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX
2	0.56488E+02	•	17	XXXXXXXXXXXXXXXX
	0.75232E+02	0.9354	9	XXXXX
	0.93976E+02	0.9524	N	XXXX
	0.11272E+03	8686.0	11	XXXXXXXXX
	0.13146E+03	8486.0	0	
	0.15021E+03	0.9932		×
	0.16895E+03	0.9932	0	
	0.18770E+03	9966.0		×
	0.20644E+03	1.0000		×

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HISTOGRAM OF FILE DK SIBOOOFF

MIN = -0.12562E+030.46875E+02 Ħ MAX

294. OF FOINTS TOTAL NUMBER POINT NORMAL IZED

> UPPER CELL BOUNDARY

FREGUENCY CUMULATIVE FREGUENCY

0.0034 0.0034

-0.11484E+03

0.0068 -0.10406E+03

0.0170 **0.006** -0.93280E+02 -0.82498E+02

0.0374 -0.71717E+02 -0.60936E+02 -0.50155E+02

0.0612 0.1088 0.1667 -0.39374E+02

0.3435 0.2381 -0.28593E+02 -0.17812E+02 -0.70307E+01

0.5238 0.37504E+01 0.14531E+02

0.25313E+02

0.6905 0.8095 0.9082

1,0000 0.36094E+02 0.46875E+02

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F-4185C -1-R100

HISTOGRAM OF FILE DK F3B0000FF

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294. TOTAL NUMBER OF FOINTS

FREGUENCY POINT NORMALIZED CUMULATIVE FREQUENCY UPPER CELL BOUNDARY

-0.33075E+02

0.0034 0.0238 0.0748 0.1599 -0.27563E+02 -0.22051E+02

0.2483

0.5000 0.3741 -0.16539E+02 -0.11027E+02 -0.55152E+01

0.5986 0.7415 -0.32043E-02 0.11021E+02 0.55088E+01

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0.8435 0.16533E+02

0.9082 0.22045E+02 0.27557E+02 0.33069E+02

0.9422 0.9898 0.9932 0.38581E+02 9966.0 1.0000

0.44093E+02

0.49605E+02

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FREQUENCY FOINT NORMAL I ZED CUMULATIVE UPPER CELL BOUNDARY

FREGUENCY 0.0068

0.0238 0.0782 -0.29678E+02 -0.25107E+03 -0.20535E+02 -0.15964E+02

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xxxxxxxxxxxxxxxxxxxxxxxxxxxxxxx XXXXXXXXXXXXXXXXXXXXXXXXX XXXXXXXXXXXXXXXXXXXXX XXXXXXXXXXXXXXXXX xxxxxxxxxxxx 18 25 0.3027

-0.11392E+02

-0.68209E+01 -0.22495E+01 0.23220E+01

36 37 0.5510 0.4252

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0.6429 0.7585 0.9422 0.8741

0.11465E+02 0.68935E+01

0.9626 0.16037E+02 0.2060BE+02 0.25180E+02

0.9830 0.34323E+02 0.29751E+02

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0.38894E+02

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NORMALI CUMULAT FREQUEN	0.0034	0.0102	0.0102	0.0136	0.0272	0.0442	0.1054	0.2007	0.3469	0.4388	0.5714	0.6735	0.7891	0.8980	0.9728	1.0000
UPPER CELL Boundary	0.38367E+02	0.34315E+02	0.30263E+02	0.26211E+02	0.22159E+02	0.18106E+02	0.14054E+02	0.10002E+02	0.59498E+01	0.18976E+01	0.21546E+01	0.6206BE+01	0.10259E+02	0.14311E+02	0.18363E+02	0.22416E+02
		LL NORMALIZED CUMULATIVE FREGUENCY +02 0.0034	NORMALIZED FOINT CUMULATIVE FREQUEN FREQUENCY 2 0.0034 1	NORMALIZED FOINT CUMULATIVE FREQUEN FREQUENCY 2 0.0034 1 2 0.0102 2	NORMALIZED FUINT CUMULATIVE FREQUEN FREQUENCY 2 0.0034 1 2 0.0102 2 2 0.0102 0	NURMALIZED FUINT CUMULATIVE FREQUEN FREQUENCY 2 0.0034 1 2 0.0102 2 2 0.0102 0 2 0.0136 1	NURMALIZED FUINT CUMULATIVE FREQUENCY FREQUENCY 2 0.0034 1 2 0.0102 2 2 0.0102 0 2 0.0136 1 2 0.0272 4 5	NORMALIZED FOINT CUMULATIVE FREQUENC FREQUENCY 2 0.0034 1 2 0.0102 2 2 0.0102 0 2 0.0136 1 2 0.0272 4 2 0.0442 5 5 0.0442 5	NURMALIZED FUINT CUMULATIVE FREQUENCY FREQUENCY 2 0.0034 1 2 0.0102 2 2 0.0102 0 2 0.0136 1 2 0.0272 4 2 0.0442 5 2 0.1054 18	CUMULATIVE FREQUENT FREQUENCY 2 0.0034 1 2 0.0102 2 2 0.0102 0 2 0.0136 1 2 0.0272 4 2 0.042 5 2 0.042 5 1 0.3469 43	CUMULATIVE FREQUENT FREQUENCY 2 0.0034 2 0.0102 2 0.0102 2 0.0136 2 0.0136 2 0.0272 2 0.042 2 0.0442 5 0.0442 1 0.3469 1 0.3469 1 0.4388	CUMULATIVE FREQUENT FREQUENCY 02 0.0034 1 02 0.0102 2 02 0.0102 0 02 0.0136 1 02 0.0272 4 02 0.0272 4 02 0.0442 5 02 0.2007 28 01 0.3469 43 01 0.4388 27	CUMULATIVE FREQUENT FREQUENCY 2 0.0034 2 0.0102 2 0.0102 2 0.0136 2 0.0136 2 0.0272 2 0.0272 3 0.0442 5 0.1054 1 0.3469 1 0.3469 1 0.5714 1 0.5714 3 30	CUMULATIVE FREQUENT FREQUENCY COMPLETIVE FREQUENCY 1 0.0034 1 0.00136 0.0136 0.00272 0.0272 0.0272 0.02072 0.02072 0.02469 0.03469 0.0	CUMULATIVE FREQUENT FREQUENCY CONOCA 1 1 2 0.0102 2 0.0102 2 0.01036 1 1 2 0.0272 4 4 2 0.0272 28 1 18 0.3469 43 1 0.3469 1 0.5714 33 1 0.5714 1 0.571	CUMULATIVE FREQUENT FREQUENCY CONOCATOR TO 0.0034 1 2 0.0102 0 0 0.0136 1 1 0 0.0272 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0

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-0.43324E+02 ŧ ZIX 0.30587E+02 H MAX

294. TOTAL NUMBER OF FOINTS

FREGUENCY FOINT NORMAL I ZED CUMULATIVE FREGUENCY UPPER CELL BOUNDARY

0.0034 0.0034 0.0136 -0.34085E+02 -0.29466E+02 -0.38705E+02

0.0170 -0.24846E+02

0.1122 0.0544 -0.20227E+02

-0.15607E+02 -0.10988E+02 -0.63684E+01

0.3163 0.4218 0.2007 -0.17489E+01

.5680 0.7075 0.8265 0.8912 0.12109E+02 0.16729E+02 0.28706E+01 0.74900E+01

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0.21348E+02 0.2596BE+02 0.30587E+02

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APPENDIX VI

HISTOGRAMS

POLISHED-SCUFFED SPECIMENS

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SKF TECHNOLOGY SERVICES SKF INDUSTRIES, INC

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- -0.44495E102 X IX

OF POINTS TOTAL NUMBER

FREDUENCY NORMAL IZED CUMULATIVE UPPER CELL ROUNDARY

FREQUENCY 0.0068 0.0170 0.0374 0.0442 0.0646 0.1224 0.39403E102 -0.34312E402 -0.29220E402 -0.24128E402 -0.13936E102 -0.13945E401 -0.37610E101 0.13307E401

0.2619

0.5476 0.7789 0.8776 0.9286 0.9898 0.9524 0.9762 0.115146+02 0.16606E+02

0.21698E+02 0.26790E+02 0.31881E+02 0.36973E+02

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XXXX FOINT FREQUENCY 1011442422 1111174422 1111174422 HISTOGRAM OF FILE OK SZAGOOGFE NURMALIZER CUMUL ATIVE FREQUENCY 0.0034 0.0034 0.0102 0.0272 0.0772 0.1729 0.3177 0.3810 0.7891 HIN OF POINTS 0.9116 0.9490 0.9864 0.30469E+02 -0.39035E+02 -0.34401E+07 -0.29768E+92 -0.25134E+02 -0.45867E+02 -0.19867E+02 -0.19862E+01 -0.17662E+01 -0.17662E+01 -0.17652E+01 -0.11935E+02 -0.11935E+02 -0.21202E+02 -0.21202E+02 -0.21202E+02 -0.21202E+02 -0.21202E+02 TOTAL MUNEER UFFER CELL BOUNDARY Mox

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0.74173E+02 HIN = -0.96757E+02

POINT FREQUENCY NORMALIZED CUMULATIVE FREQUENCY UPFER CELL ROUNDARY

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TOTAL NUMBER

0.0034 0.0034 0.0068 0.0068 0.0102 0.0374 -0.86074E+02 -0.5391E+02 -0.54707E+02 -0.43341E+02 -0.32658E+02 -0.11292E+02 -0.11292E+02 -0.10075E+02 0.20758E+02 0.20758E+02 0.31441E+02 0.31441E+02 0.31441E+02 0.31441E+02

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0.8912

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HISTOGRAM OF FILE DK S4A0000FF

Z 0.75571E+02

OF POINTS TOTAL NUMBER NORMALIZED CUMULATIVE FREQUENCY UPPER CELL **BOUNDARY**

122 222 223 237 237 24 100 3 -0.56816E+02 -0.37164E+02 -0.37164E+02 -0.21513E+02 -0.12613E+02 -0.38611E+01 0.49646E+01 0.13790E+02 0.22616E+02 0.231442E+02 0.40268E+02 0.40268E+02 0.40268E+02 0.40268E+02

0.0034 0.0102 0.0510 0.2109 0.5068 0.6068 0.6053

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SKF TECHNOLOGY SERVICES SKF INDUSTRIES, INC

VI - 5

FILE DK F6A0000FF 6 HISTOGRAM

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MIN = -0.37654E+02 0.49389E+02 ij

OF FOINTS TOTAL NUMBER

POINT FREQUENCY NORMALIZED CUMULATIVE FREQUENCY UFFER CELL BOUNDARY

XXXXXXXXXXXXXXXXXXXXXXXXXXXXX XXXXXXXXXXXXXXXXXXXXXXXX XXXXXXXXXXX XXXXXXXXXX XXXXXXXXXXX 111123334554111 0.0374 0.1190 .4592 -0.32214E+02 -0.26773E+02 -0.21333E+02

-0.15893E+02 -0.10453E+02 -0.10453E+02 0.42759E+00 0.58678E+01 0.11308E+02 0.22188E+02 0.27628E+02 0.37628E+02 0.37628E+02 0.37628E+02 0.37628E+02 0.47389E+02

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294.	FOINT FREQUENCY	-	₹	œ	16	21	42	0E			25	13	12	^	m	ທ	•
	NORMALIZED CUMULATIVE FREQUENCY	0.0034	0.0170	0.0442	9860.0	0.1701	0.3129	0.4150	0.5510	0.7687	0.8537	0.8980	0.9388	0.9626	0.9728	0.9898	
	UFFER CELL. BOUNDARY	-0.56290E+02	-0.47851E+02	-0.39411E+02	-0.30972E+02	-0.22532E+02	-0.14093E+02	-0.56532E+01	0.27864E+01	0.11226E+02	0.19665E+02	0.28105E+02	0.36544E+02	0.44984E102	0.53424E+02	0.61863E+02	

VI - 7

AT83D013

APPENDIX VII

HISTOGRAMS

POLISHED-UNSCUFFED SPECIMENS

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HISTOGRAN OF FILE DK F2R0000FF

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[= 0.35580E+02 HIN = -0.11786E+03

FOINT FREQUENCY 294. NORMAL I ZED CHMULAT I VE FREQUENCY OF POINTS 0.0034 0.0102 0.0102 0.0102 0.0102 0.0204 0.0238 0.0272 0.0374 0,10827E+03 0,9867RE+02 0,8998E+02 0,69908E+02 0,6318E+02 0,31549E+02 0,31549E+02 0,2179E+02 0,2179E+02 0,2179E+02 0,2179E+02 0,21794E+01 0,68104E+01 0,68104E+01 0,68104E+01 0,25990E+02 IUTAL MUMBER UFFER CELL BOUNTARY

HISTOGRAM OF FILE DK S2B0000FF

F-4185C-1-R100

X = 0,30696EF02 MIN = -0,83453EF02

FOINT FREQUENCY NORMALIZED CUMULATIVE FREQUENCY OF FOINTS 0.0068 0.0068 0.0136 0.0138 0.0238 0.0476 0.0884 0.1293 0.7993 0.9252 0.9592 1.0000 0.69184E+02 0.62050E+02 0.5791E+02 0.4781E+02 0.3547E+02 0.26378E+02 0.19244E+02 0.19244E+02 0.15110E+02 0.21591E+01 0.92934E+01 0.92934E+01 0.2562E+02 0.2562E+02 0.2562E+02 TOTAL NUMBER UPPER CELL ROUNDARY

VII-3

SKF TECHNOLOGY SERVICES SKF INDUSTRIES, INC

HISTOGRAM OF FILE DE FABOOOFF

- 0.18843E+02 MIN = -0.40320E+02

TOTAL NUMBER OF PUINIS = 294.

FREQUENCY × NORMALIZED CUMULATIVE FREQUENCY 0.0034 0.0068 0.0068 0.0068 0.0068 0.5646 0.1531 0.3231 0.9694 -0.32925E+02 -0.29227E+02 -0.21832E+02 -0.118134E+02 -0.10739E+02 -0.70412E+01 -0.33435E+01 0.35423E+00 0.40519E+01 0.11445E+02 0.11445E+02 UPPER CELL ROUNDARY

F-4185C-1-R100

SKF TECHNOLOGY SERVICES SKF INDUSTRIES, INC

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-0.38842E+02 0.23629E+02 HIN

FOINTS TOTAL NUMBER

FOINT FREQUENCY UPFER CELL ROUNDARY

NURMALIZED CUMULATIVE FREQUENCY 0.0000 0.0068 0.0176 0.0172 0.0782 0.1088 0.1701 0.3027 0.27129E+02 0.23224E+02 0.15312E+02 0.115415E+02 0.115415E+02 0.20218E+00 0.20218E+00 0.41066E+01 0.41066E+01 0.11915E+02 0.15820E+02 0.1915E+02 0.19324E+02 -0.34938E+02 -0.31033E+02

0.6871

0,8912

VII-5

SKF TECHNOLOGY SERVICES SKF INDUSTRIES, INC

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-0,21880E+02 ZIE 0.90113E+01

254. OF FOINTS TOTAL NUMBER

POINT FREQUENCY HORMAL IZED CUMULATIVE FREQUENCY UFFFR CFLL ROUNDARY

XXXXXXXXXXXXXX 1234564933224 0.0034 0.0170 0.0238 0.0306 0.0408 0.0034 0.0884 0.1429 -0.18019E+02 -0.14157E+02 -0.19226E+02 -0.10296E+02 -0.83650E+01 -0.6433E+01 -0.45036E+01 -0.25729E+01

VII-6

0.2415

0.7517

0.12885E+01 0.32192E+01 0.51499E+01 0.70806E+01 0.90113E+01

TOTAL NUMBER OF POINTS = 294.

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BOUNDARY	CUMULATIVE	FREQUENCY	NCY
-0.29349E+02	0.0034	~	×
-0.26054E102	0.0034	0	
-0.22760E+02	_	0	
-0.19465E+02	0.0048	-	· *
-0,16170E+02	0.0340	80	XXXXXX
-0,12875E+02	0.0578	7	XXXXXX
-0.95804E+01	0.1156	17	XXXXXXXXXXXXX
-0,62856E+01	0.1837	20	XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX
-0.2990BE+01	0.2959	33	XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX
0.30405E+00	_	61	**************************************
0.35989E+01	0.6599	46	********
0.68937E+01	0.8435	ņ	**************************************
0.10188E+02	0.9524	32	**************************************
0.13483E+02	0.9864	10	XXXXXXX
0.16778E+02	0.9966	P *)	XXX
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APPENDIX VIII

SCUFFING TEST PROCEDURE

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APPENDIX VIII

SCUFFING TEST PROCEDURE

- 1. The tests are conducted on the Geared Roller Test Machine (Figure A-1) at the following conditions:
 - a) Lubricant Bulk Temperature 180°F
 - b) Load Incremented 25 lbs/5 min.
- 2. Test Method Pre-Start

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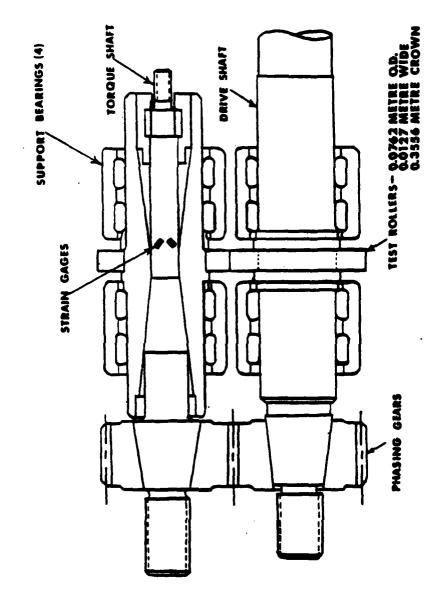
- a) Charge system with specified lubricant.
- b) Install test rollers (odd # on upper shaft and even # on lower shaft) with numbers facing the geared portion of the shaft.
- c) Obtain the required 180°F bulk lubricant temperature.
- d) Calibrate the load indicator as required prior to start.
- e) Calibrate Strain Sert to zero.
- f) Check operation of thermocouple on upper and lower rollers.
- g) Apply air pressure as set on gage to 65 lbs maximum.

FIGURE A-1

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GEARED ROLLER TEST MACHINE SCHEMATIC



3. Test Method - Procedure

AT83D013

- a) Reset safety switches to on positions and start system.
- b) Start stopwatch and run for 15 minutes at tare load.
- c) While at tare load record the following:
 - (1) Torque at start of load cycle.
 - (2) Upper and lower roller temperature.
 - (3) Bulk oil temperature.
 - (4) Load air pressure.
 - (5) Pump and inlet oil pressures.
 - (6) Torgue at end of load cycle.
- d) Increase load by 25 lb and record the torque.
- e) At the three minute mark record the following:
 - (1) Upper and lower roller temperature.
 - (2) Bulk oil temperature.
 - (3) Load air pressure.
 - (4) Pump and inlet oil pressures.
 - (5) Torque at the end of the load cycle.

- f) Repeat until a failure is obtained or:
 - (1) Torque increases by 100 units over the previous load point.
 - (2) Torque limit of 950 units is reached.
- 4. Test Method Unload

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- a) In some cases it may be necessary to unload the system prior to obtaining a failure. In case of such an event, the following procedure is to be followed:
 - (1) Reduct load by 50 lb/min and record torque and roller temperature.
 - (2) On reaching tare load operate the system for an additional fifteen minutes.
 - (3) Turn the system off allowing the lubricant to circulate.
- 5. Test Method Reload
 - a) After an unload as described in 5, the following procedure is to be followed to reload the system:
 - (1) Reach a bulk oil temperature of 180°F.
 - (2) Calibrate all systems as required.

- (3) Turn system on and operate at tare load for fifteen minutes.
- (4) Begin load in 100 lb increments every 2 minutes until a 50 lb less than the load at which shutdown occurred.
- (5) Begin procedure as outlined in (4). (Note: Record required data at each load point).
- 6. System Flush From time to time the operating fluid in the system will be varied. In such an event the following procedure should be followed in order to insure of no contamination.
 - a) Drain system including all lines.
 - b) Fill sump with approximately 2 gallons of hexane and start pump and operate 10 minutes.
 - c) Rinse all parts in hexane.

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- d) Drain system and add 2 gallons Hercolube A.
- e) Start pump operate for 10 minutes.
- f) Drain system and add 1.5 gallons of test fluid and rinse all other parts in test fluid remove filter element.
- g) Operate pump for 10 minutes.

h) Drain system - add new filter element - fill sump with test fluid to level of heater elements.

4. (6)